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I put a lot of time into producing these files which is why you are met with this page when you open the file.

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It is my hope that you find the file of use to you personally – I know that I would have liked to have found some of these files years ago – they would have saved me a lot of time !

Colin Hinson

In the village of Blunham, Bedfordshire.



*H. J. Tyson*

**RESTRICTED**



**BASIC PRINCIPLES AND EXAMPLES  
OF  
TIME DOMAIN REFLECTOMETRY**

**CENTRAL SERVICING  
DEVELOPMENT ESTABLISHMENT**

**ROYAL AIR FORCE  
SWANTON MORLEY  
NORFOLK.**

**RESTRICTED**

*H. J. Tyson*



## INTRODUCTION

1. The information contained in the following note is additional technical information to supplement the Hewlett Packard Handbooks issued with the 10S/4935-99-956-8468 OSCILLOSCOPE SET CT 552.

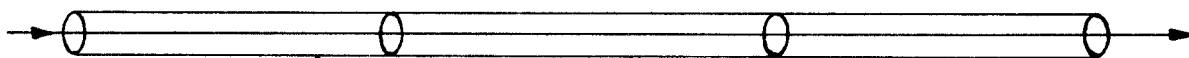
TYPICAL TDR DISPLAYS

1. Impedance Matching in co-ax. - Quarter Wave Transformers.

$Z_{in} = 50 \text{ ohms}$

$\frac{1}{4}$  wave length

$Z_{out} = 200 \text{ ohms}$



Cable Impedance =  $Z_o$ .

$Z_o = 50 \text{ ohms.}$

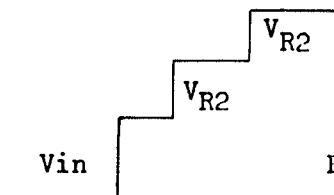
$Z_o = 100 \text{ ohms.}$

$Z_o = 200 \text{ ohms.}$

VSWR at optimum frequency may be 1.08:1

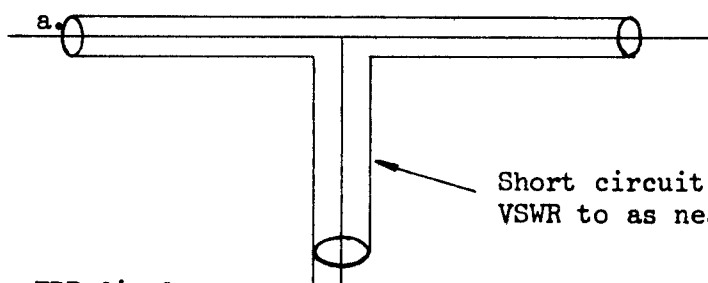
VSWR at another frequency may be 1.3:1

TDR display



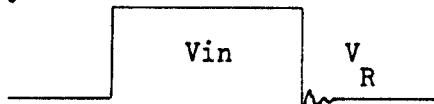
Total reflection  
co-efficient = +0.6  
Reflection co-efficient =  $\frac{V_{R1} + V_{R2}}{V_{in}}$

Tuned stubs in co-axial Lines.

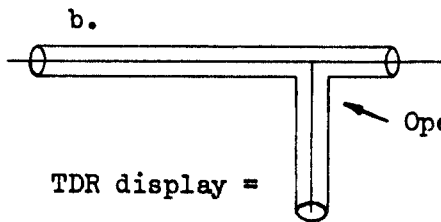


Short circuit stub used to reduce system VSWR to as near to 1:1 as possible.

TDR display =

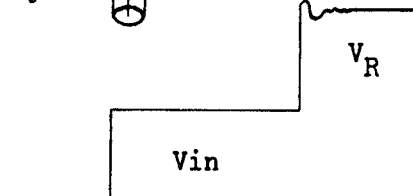


Reflection co-efficient = 1



Open stub

TDR display =

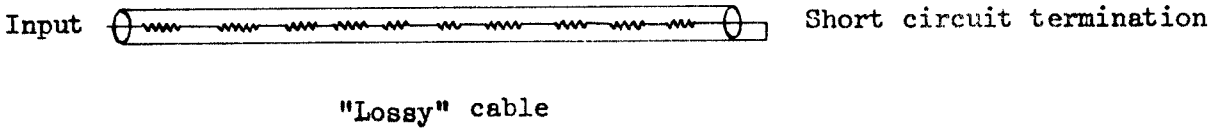


Reflection co-efficient = +1

**NOTE:** The reflection co-efficient is always 1 and cannot be altered by adjusting the length of the stubs, but the VSWR at the operating frequency can be alter ub length.

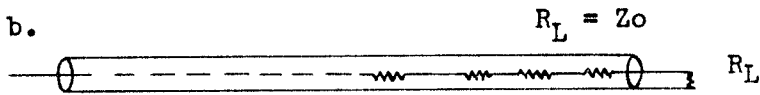
3. Attenuation (Simple)

a.



$$\text{Two way attenuation} = 20 \log \frac{V_{IN}}{V_R} = 6\text{db}$$

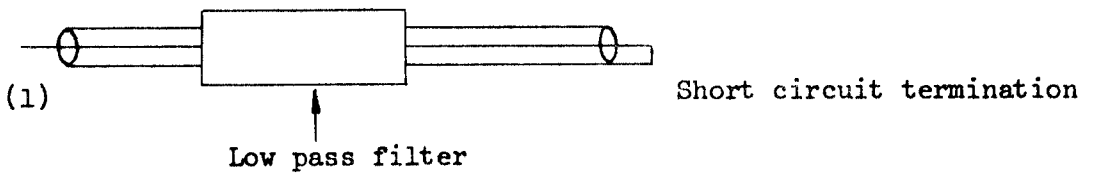
Therefore cable attenuation = 3db



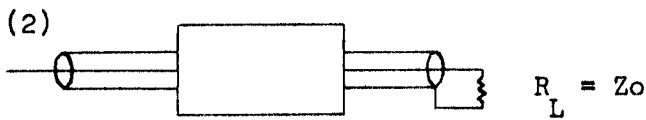
The degree of attenuation is NOT displayed.

4. Attenuation (Frequency Conscious).

a.

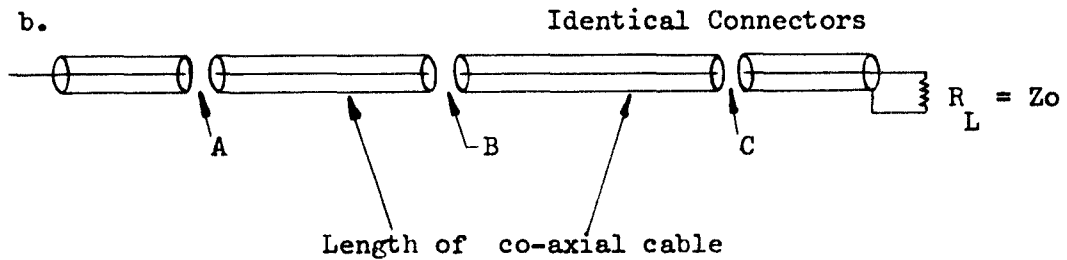


eg = 3db at 100 Mc/s and -30db at 200 Mc/s and above.



NOTE: A co-axial cable is effectively low pass filter.

eg: Transradio RG-58 c/u co-axial cable, Attenuation per 100 ft:  
 0.42 db at 1 Mc/s: 1.6 db at 10 Mc/s: 6.2 db at 100 Mc/s:  
 24 db at 1,000 M/cs: 45 db at 3,000 Mc/s.



TDR displays:

(1) reflection from A



reflection from small inductive mismatch in connector

(2) reflection from B



(3) reflection from C



Because the co-axial cable is an effective low pass filter the measurement bandwidth reduces throughout the system. Therefore the reflections from 'distant' components that cause high mismatch at SHF, may be smaller than the reflections from the strands of wire of the outer brading of the co-axial at the input end of a system.

5. The comments in this note should not be taken as criticisms of the Time Domain Reflectometer as a test instrument. It is a very effective instrument when used on LF to UHF co-axial systems. This note illustrates some of the differences between broad band and discrete frequency testing.

## SOME THEORETICAL TIME DOMAIN REFLECTOMETER DISPLAYS

1. The Characteristic Impedance of any cable, can be computed in terms of the voltage of the step appearing on the TDR oscilloscope before, and after the cable has been connected, ie:

If:  $E_1$  = Incident voltage of step

$Z_1$  = Characteristic Impedance of cable

$Z_0$  = Output Impedance of Step generator

$E_s$  = Voltage of step source.

Then:  $E_1 = \frac{Z_L E_s}{Z_0 + Z_L}$  (See figs 1)

2. The incident step voltage is propagated down the co-axial cable and will be reflected from any discontinuities and from the cable termination.

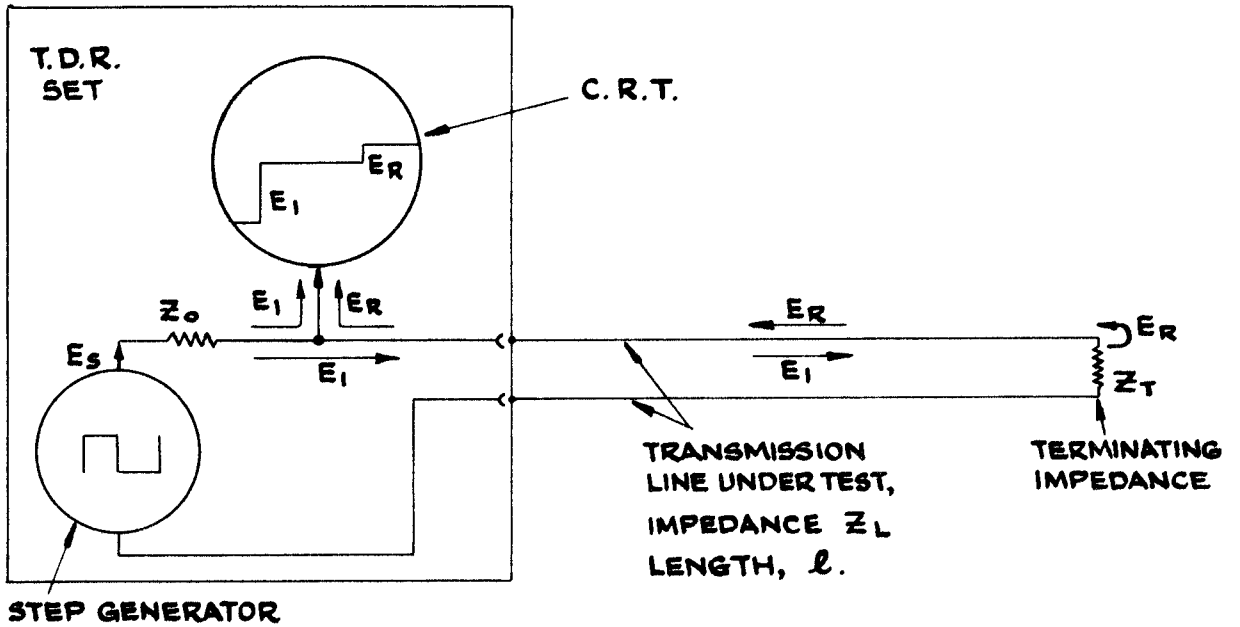
The magnitude shape and sense of the reflection will depend on the nature of the discontinuities and will appear on the display screen of the TDR set, delayed from the incident step by a time equal to the propagation delay of the co-axial cable from the source to the discontinuity and return (Return path delay). The reflection is added algebraically to the incident step voltage. (fig 2).

3. If the co-axial cable is terminated by an open circuit the reflected step voltage will be equal in amplitude to the incident voltage and will be positive. If terminated by a short circuit the reflection will be negative.

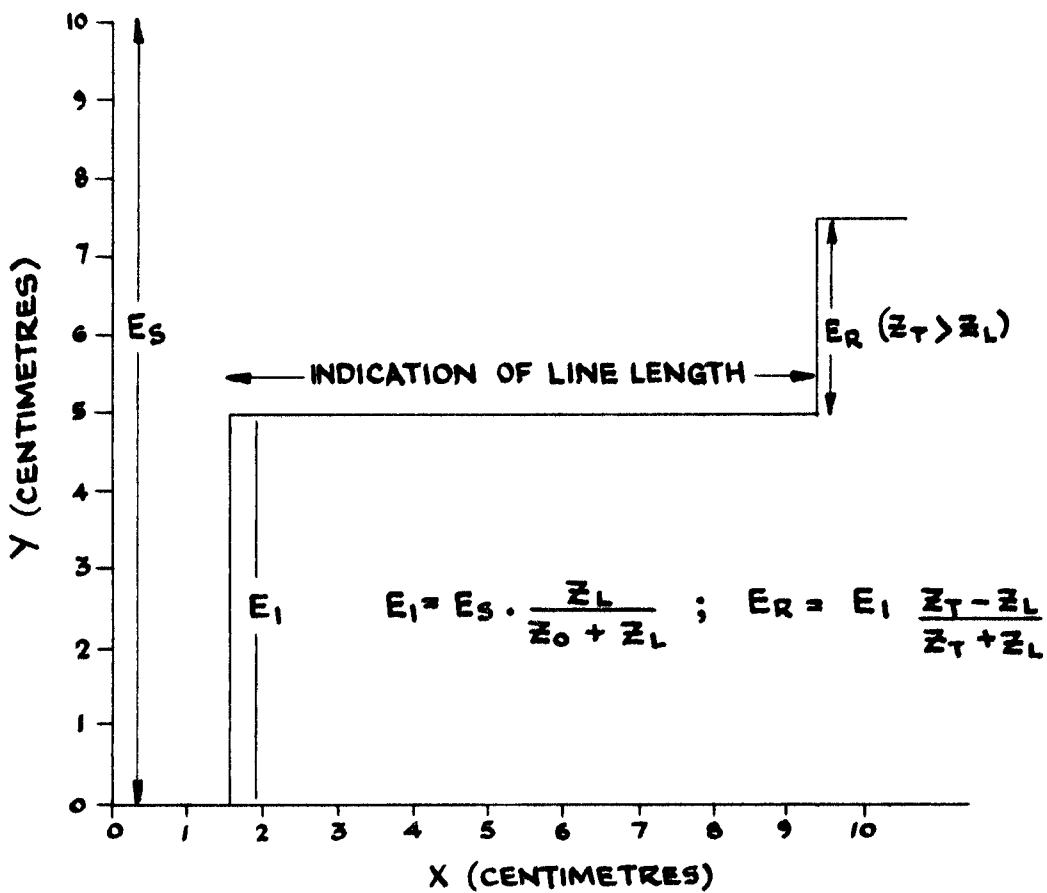
If the termination is an impedance matching that of the cable at all frequencies, no reflection will result. The position of a perfectly matched termination will not be apparent. (fig 3).

4. Assume a cable with a perfectly matched load, but somewhere along its length is a component causing a small standing wave ratio (SWR), such as a co-axial plug/socket interconnection, this will cause a proportion of the incident step to be reflected back, and will show on the TDR screen as a small pulse, either positive or negative depending on whether it is caused by series Inductance or parallel capacitance. The amplitude of the reflection is proportional to the degree of SWR. (fig 4).

5. In the case of a resonant aerial, the arrival of the step at the aerial, will excite it into resonance. This resonance will be transmitted back to the TDR and displayed as an exponentially decaying sine wave. If the aerial has an impedance matching transformer, it will appear as a short circuit, after the disturbance due to its resonant excitation; if it is fed directly to the aerial, it will appear as an open circuit. (fig 5).



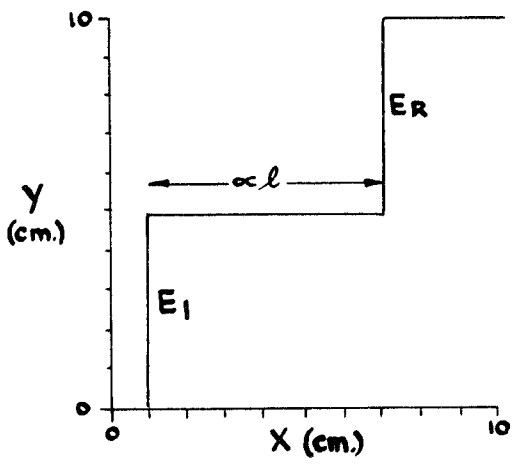
TIME DOMAIN REFLECTOMETER ~ BLOCK DIAGRAM. FIG. 1A.



T. D. R. DISPLAY.

FIG. 1B.





C.R.T. DISPLAY

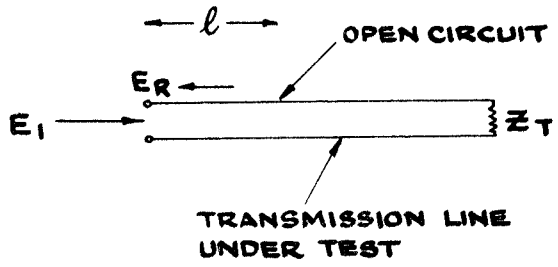


FIG. 2.

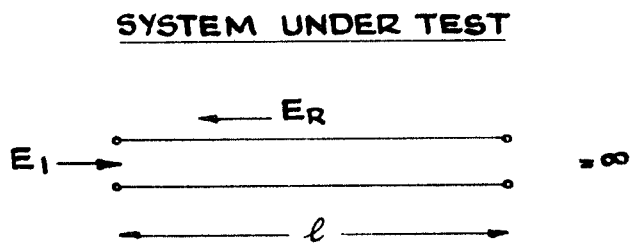
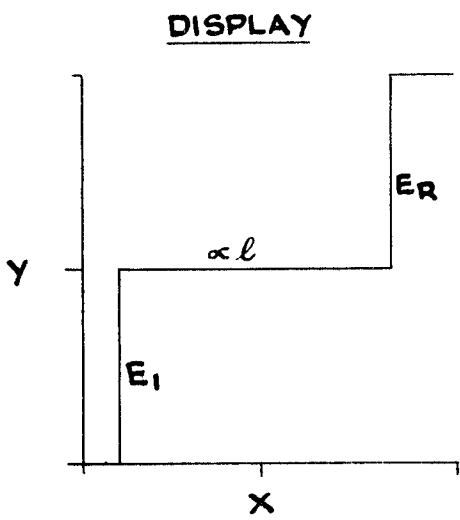


FIG. 3A.

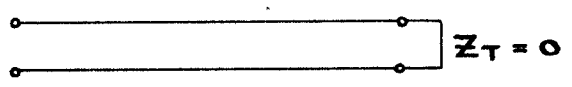
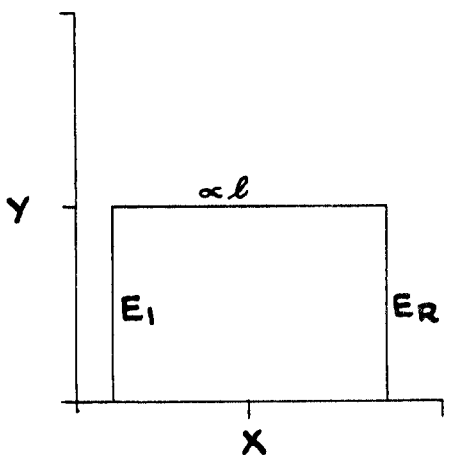
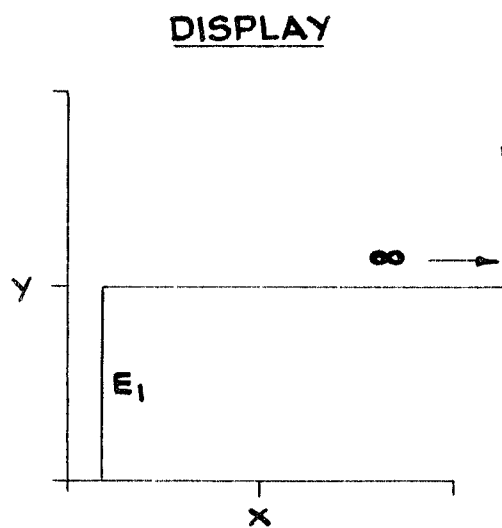


FIG. 3b.



SYSTEM UNDER TEST

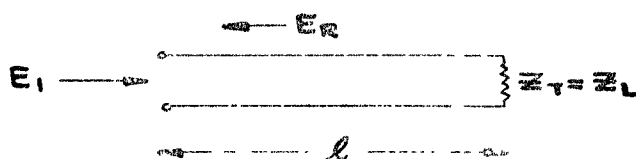


FIG. 3c

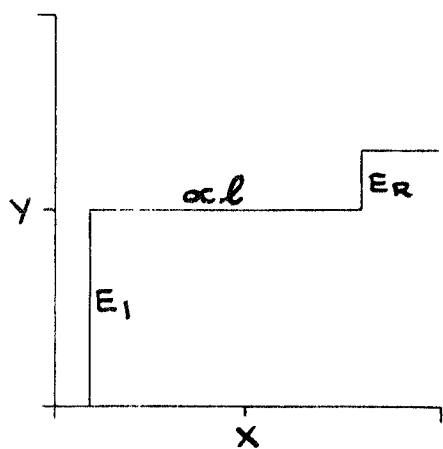


FIG. 3D

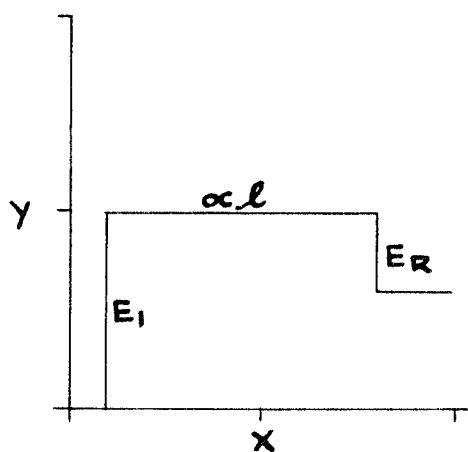
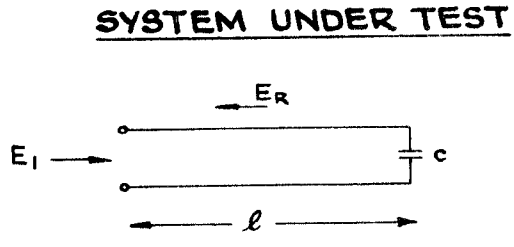
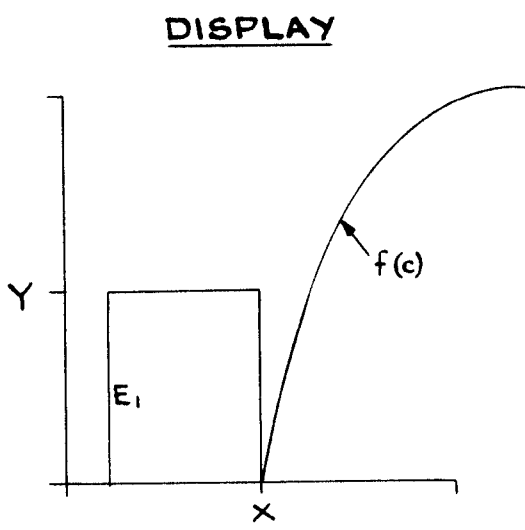
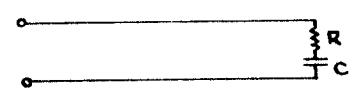
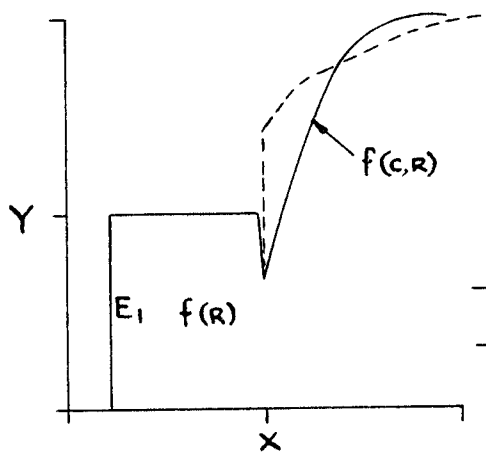


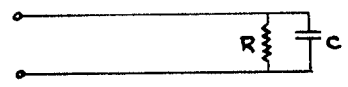
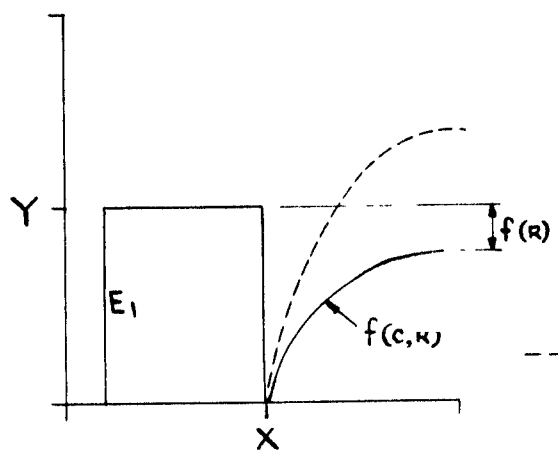
FIG. 3E



**FIG. 3 F.**



**FIG. 3 G.**



**FIG. 3 H.**

DISPLAY

SYSTEM UNDER TEST

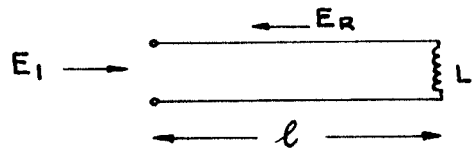
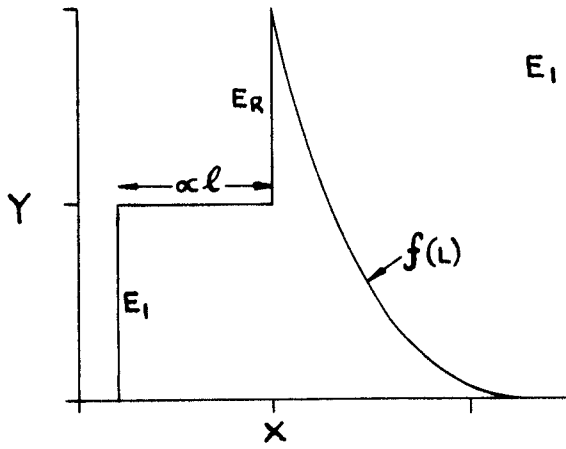


FIG. 3J

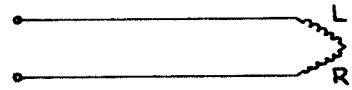
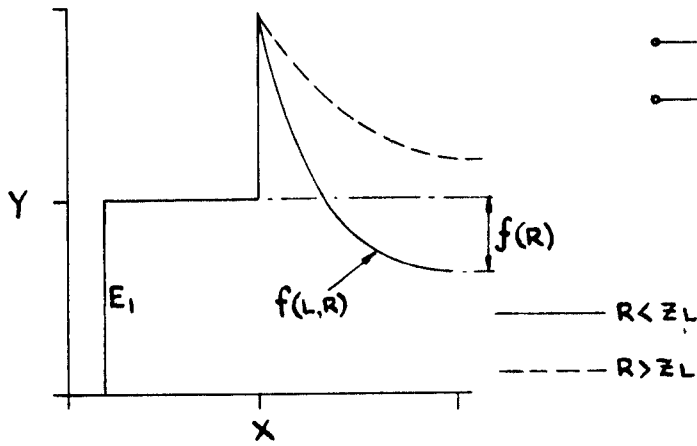


FIG. 3K

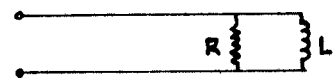
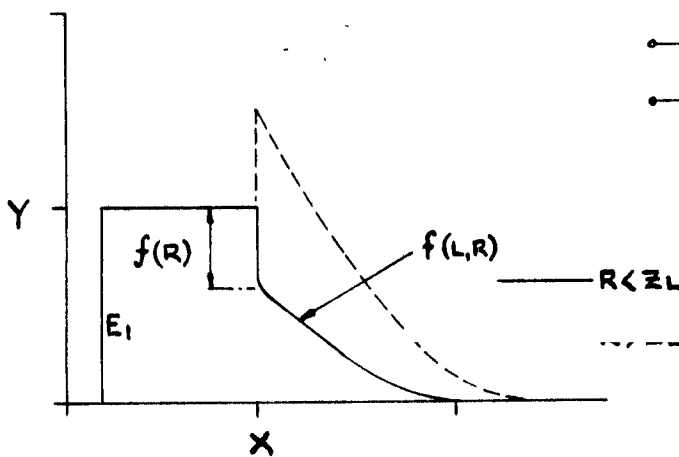


FIG. 3L

DISPLAY

SYSTEM UNDER TEST

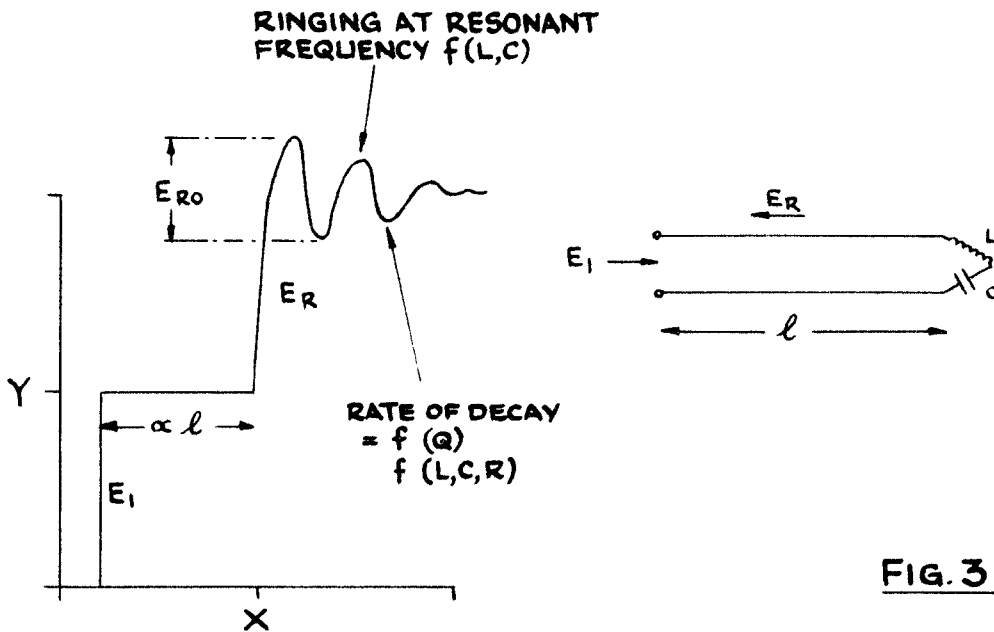


FIG. 3M

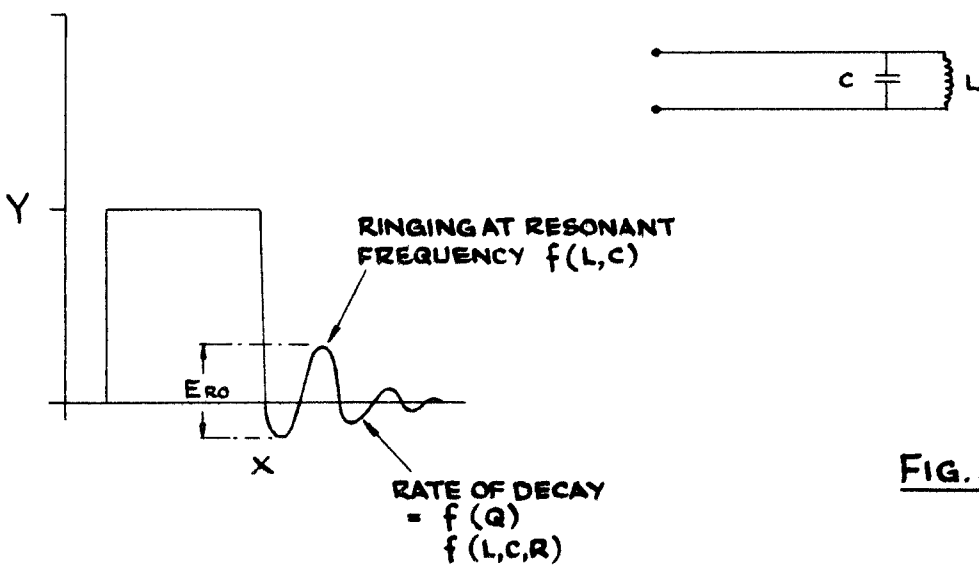


FIG. 3N.

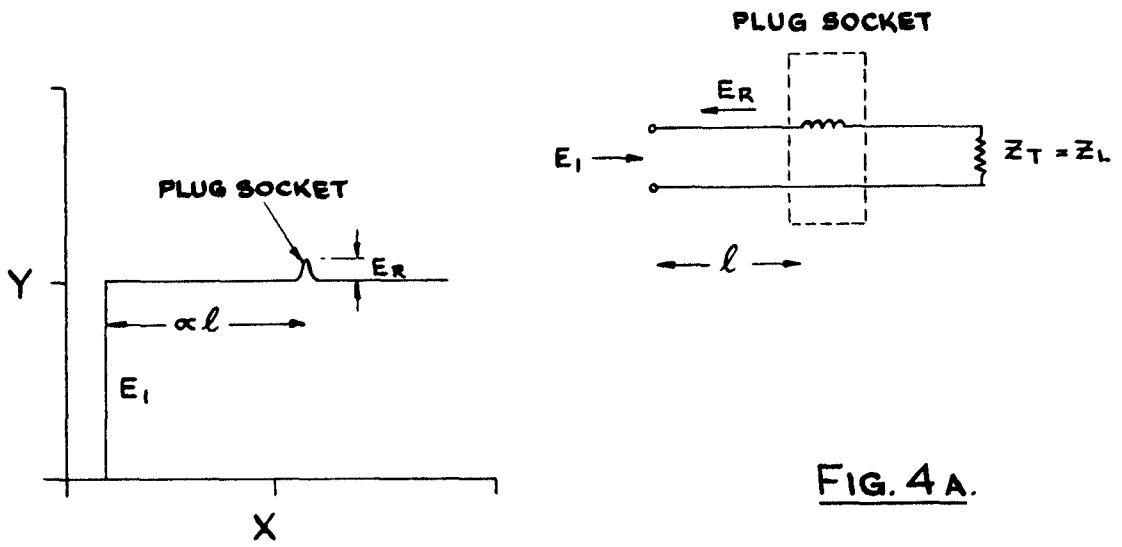
THE AMPLITUDE OF THE RESONANT OSCILLATIONS,  $E_{Ro}$  IS DEPENDANT UPON:-

RESONANT FREQUENCIES, CIRCUIT IMPEDANCES, RESONANT CIRCUIT 'Q',  
CABLE CHARACTERISTICS, RISE TIMES, ETC.

MATHEMATICAL EQUATIONS FOR THE VARIOUS FUNCTIONS CAN BE FOUND IN:-

HEWLETT - PACKARD - APPLICATION NOTE 62  
TIME DOMAIN REFLECTOMETRY



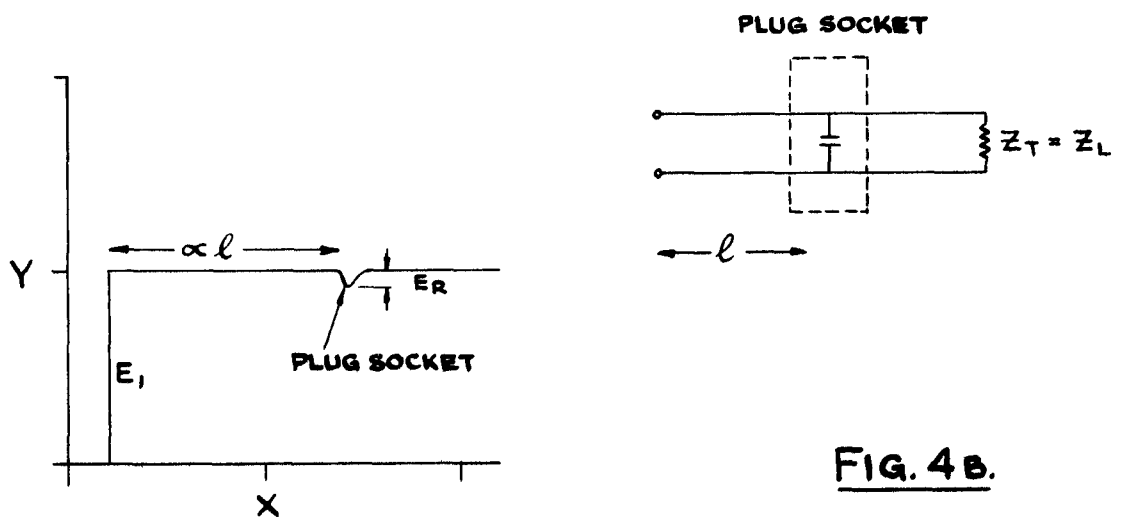


**FIG. 4A.**

REFLECTION COEFFICIENT,  $\rho$ :-  $\rho = \frac{E_R}{E_I} = \frac{Z_T - Z_L}{Z_T + Z_L}$

STANDING WAVE RATIO (S.W.R.):-

$$\text{S.W.R.} = \frac{1 + (\rho)}{1 - (\rho)}$$



**FIG. 4B.**

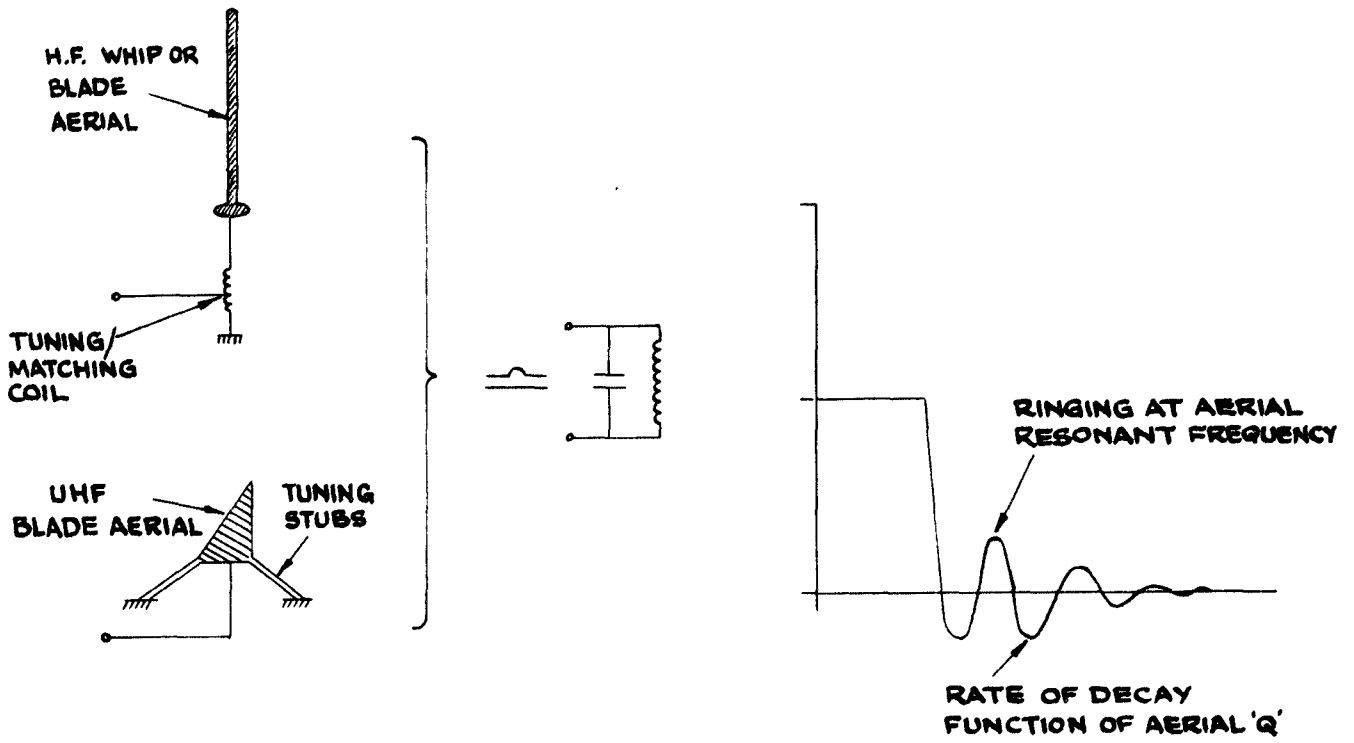


FIG. 5A.

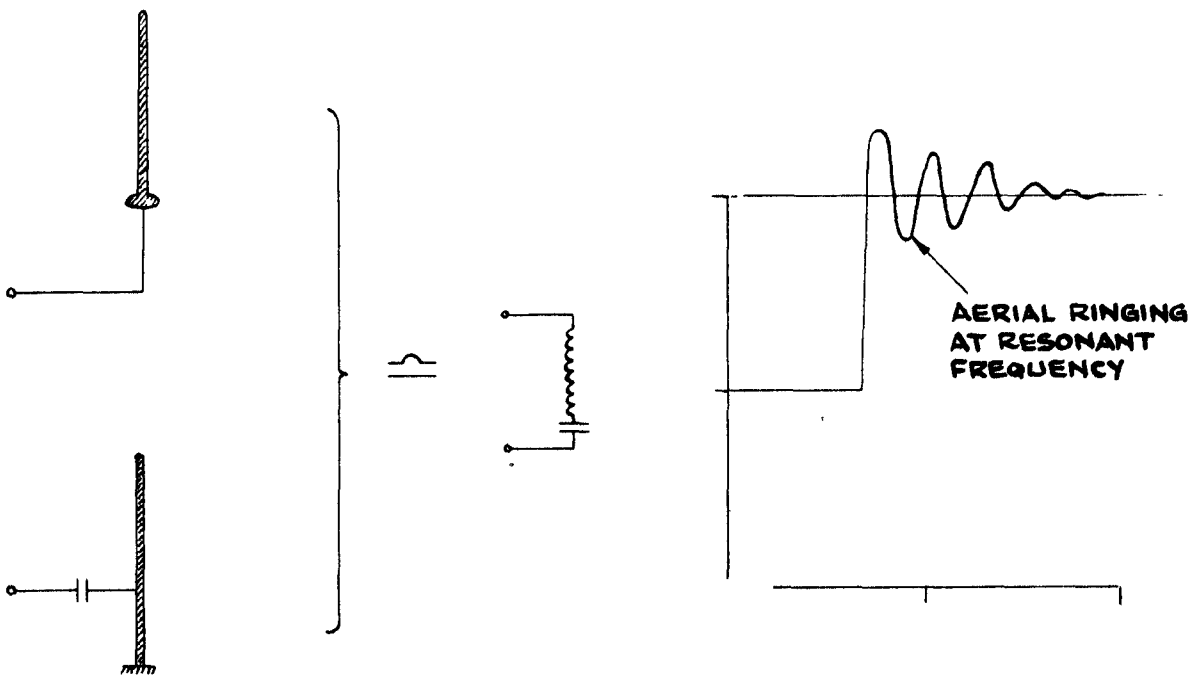
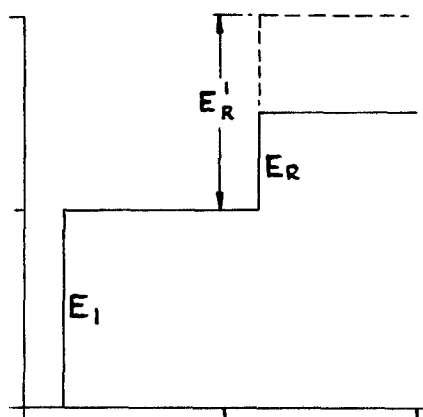


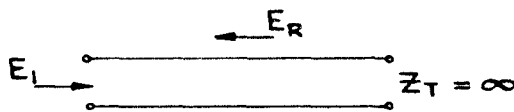
FIG. 5B.

BROAD BAND  
ATTENUATION MEASUREMENT

DISPLAY

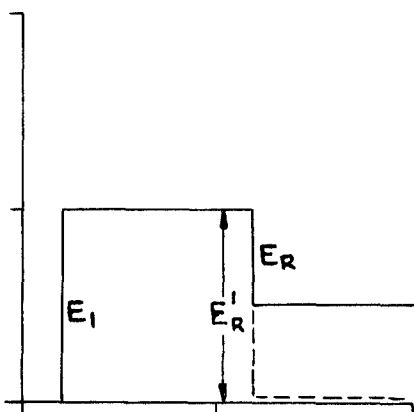


SYSTEM UNDER TEST



$$\begin{aligned} \text{ATTENUATION} &= 10 \log \left( \frac{E_R'}{E_R} \right) \\ &= 10 \log \left( \frac{E_I}{E_R} \right) \end{aligned}$$

FIG. 6 A



$$\begin{aligned} \text{ATTENUATION} &= 10 \log \left( \frac{E_R'}{E_R} \right) \\ &= 10 \log \left( \frac{E_I}{E_R} \right) \end{aligned}$$

FIG. 6 B.

BROAD BAND ATTENUATION IS CALCULATED

$$\text{FROM:- } \frac{1}{2} \left[ 20 \text{ LOG } \left( \frac{E_R'}{E_R} \right) \right] = 10 \text{ LOG } \left( \frac{E_R'}{E_R} \right)$$

WHERE  $E_R'$  IS THE VOLTAGE THAT WOULD HAVE BEEN REFLECTED THROUGH A CABLE OF ZERO LOSS, AND  $E_R$  IS THE ACTUAL VOLTAGE MEASURED.

MEASUREMENTS ARE USUALLY MADE WITH ZERO (SHORT CIRCUIT) OR INFINITE (OPEN CIRCUIT) IMPEDANCE TERMINATIONS, AS  $E_R'$  IS THEN KNOWN ( $E_R' = E_I$ )

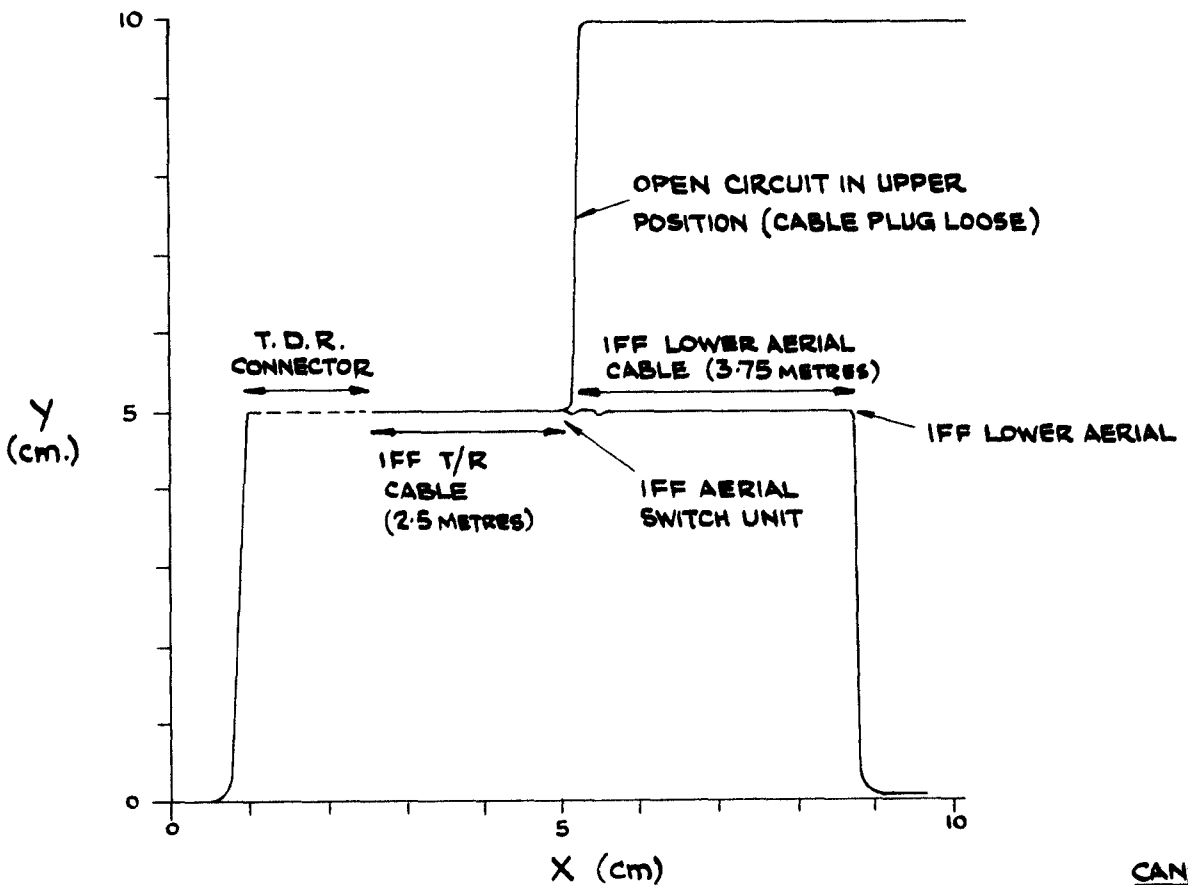
BROAD BAND ATTENUATION IS DISPLAYED AS A RESISTING TERMINATION (SEE FIGS 3c, 3d, 3e). THEREFORE LOW LOSS CABLES AND GENERAL RADIO CONNECTORS MUST BE USED TO CONNECT T M UNDER TEST (SEE ANNEX C).

TYPICAL TDR DISPLAYS IN AIRCRAFT

AIRCRAFT	FIG	OBSERVATION EQUIPMENT AND ACTION TAKEN	SERVICEABILITY
Canberra IFF	1	Open circuit at junction of the upper aerial cable and switch unit. (Plug loose)	U/S
Canberra IFF	2	Plug tightened. Result satisfactory	'S'
Lightning IFF	3	Satisfactory	'S'
Lightning IFF	4	Upper aerial gave a slightly shorter reflection fall time.	'S'
Lightning IFF	5	Slight difference was noted in switch unit upper and lower positions.	'S'
Lightning ILS	6	Localiser	'S'
Lightning ILS	7	Marker. The aerial was dismantled to observe terminating inductance, it was found to be a feature of design.	'S'
Lightning ILS	8	Glidepath	'S'
Varsity	9	Radio Compass Loop Aerial	'S'
Varsity	10	Sense aerial via filter unit (filter unit blocked the step input waveform). Unsatisfactory	-
Varsity	11	Direct to sense aerial Satisfactory	'S'
Varsity	12	Gee III	'S'
Varsity	13	Rebecca Satisfactory	'S'
Varsity	14	Rebecca Satisfactory	'S'
Varsity	15	Rebecca Satisfactory	'S'
Varsity	16	Rebecca Satisfactory	'S'
Varsity	17	VHF Satisfactory	'S'
Varsity	18	VHF Satisfactory	'S'
Britannia	19	IFF 10 Unsatisfactory	U/S
Britannia	20	IFF 10 lower aerial connected to short lead, 3db attenuation in aerial was measured.	'S'
Britannia	21	Serviceable aerial connected to cable, tested.	'S'
Britannia	22	Serviceable aerial connected to aircraft	'S'
Britannia	23	Aerial switch to flight position. Display indicated that at some part of the switching cycle the two _____ parallel. Lower aerial contacts were found not to be opening.	U/S
Britannia	24	VHF aerial	'S'
Britannia	25	STR 18 Aerial system Port High Z termination	U/S

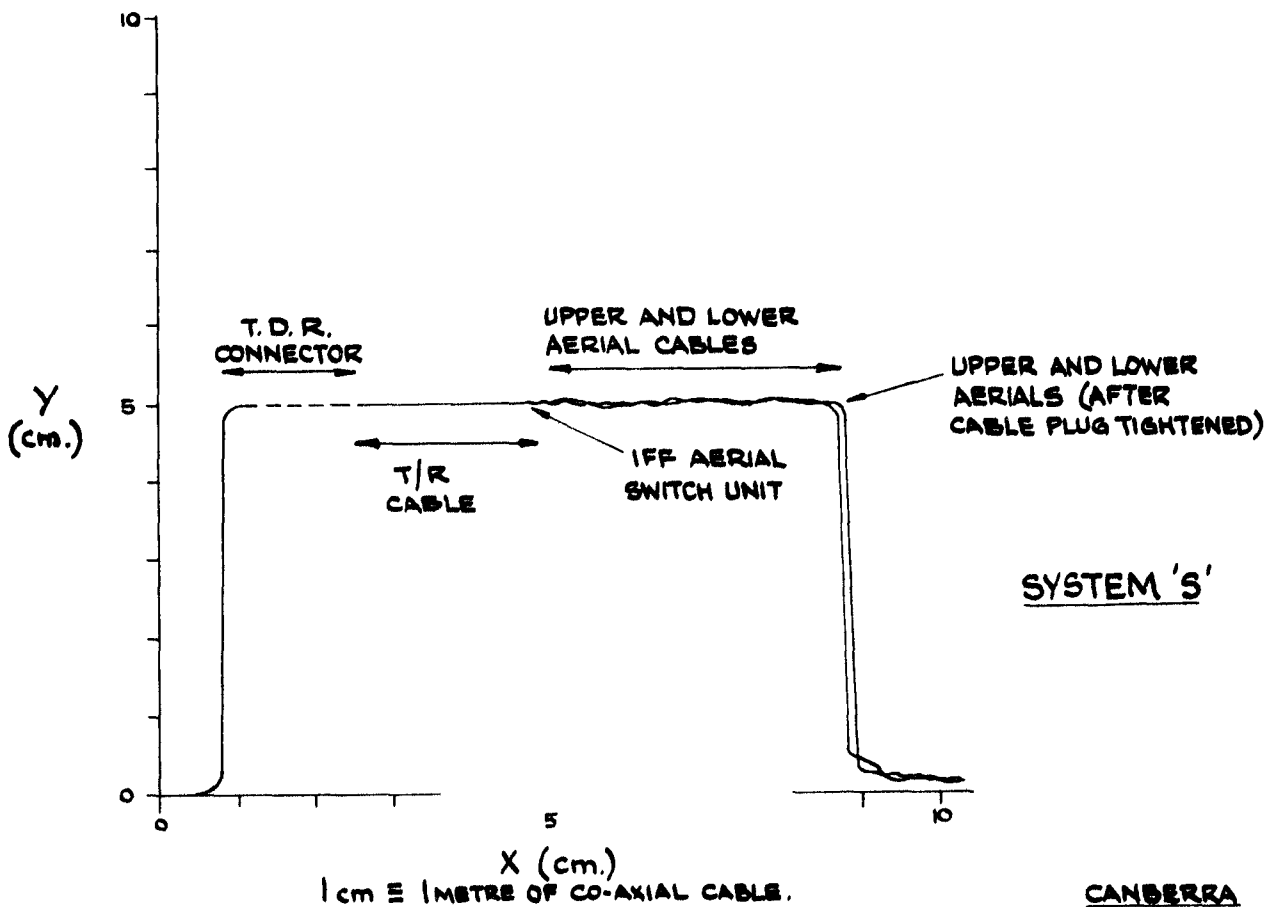
AIRCRAFT	FIG	OBSERVATION EQUIPMENT AND ACTION TAKEN	SERVICEABILITY
Britannia	26	STR 18 Aerial System	'S'
Britannia	27	STR 18 Aerial Wires Port Hing & Termination	U/S
Britannia	28	STR 18 Aerial Wires	'S'
Comet	29	ILS Glide Path	'S'
Comet	30	Localiser	'S'
Comet	31	Marker	'S'
Comet	32	UHF	'S'
Vulcan	33	IFF 10 Aerial System lower Aerial	'S'
Vulcan	34	IFF 10 Upper and Flight positions	U/S
Vulcan	35	IFF Satisfactory Display	'S'
Vulcan	36	IFF Satisfactory Display	'S'
Vulcan	37	IFF Satisfactory Display	'S'





1 cm = 1 METRE OF CO-AXIAL CABLE.  
 N.B. THE ACTUAL T.D.R. CONNECTOR LENGTH WILL DEPEND ON ACTUAL CABLE IN USE

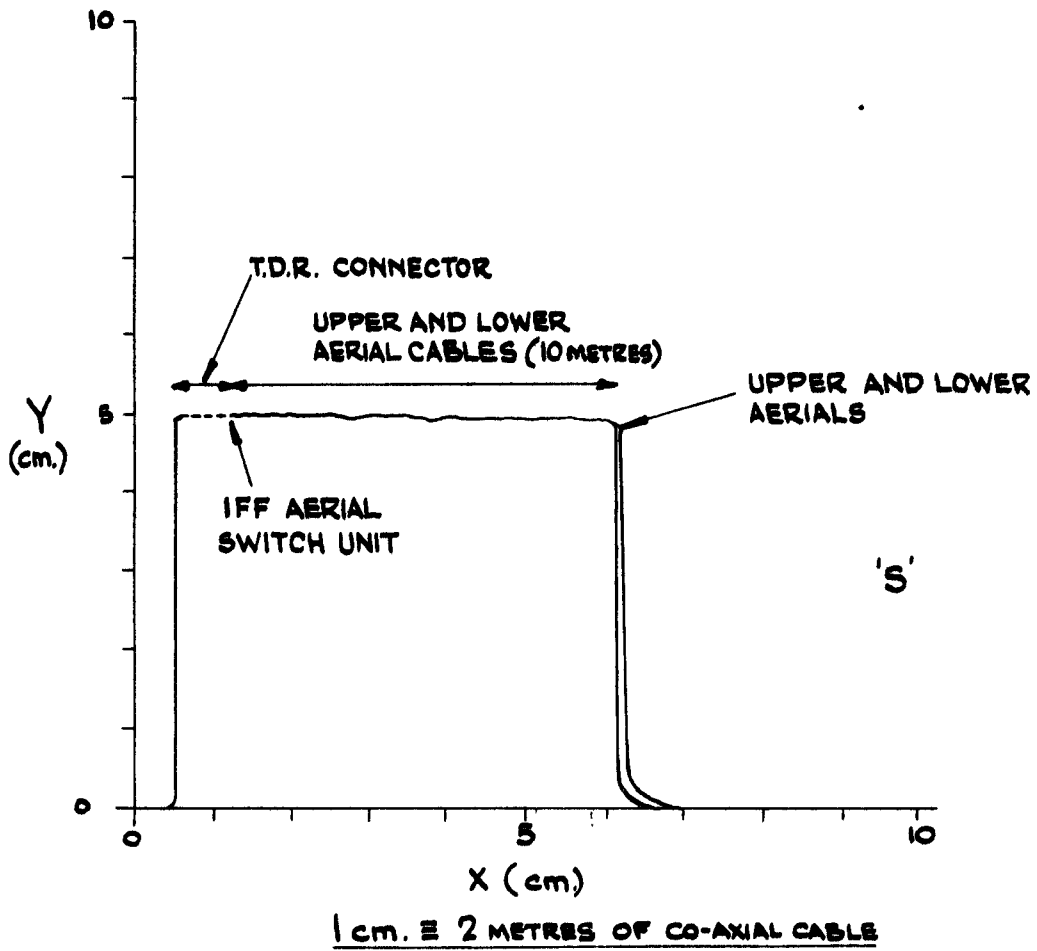
CANBERRA  
FIG. 1.



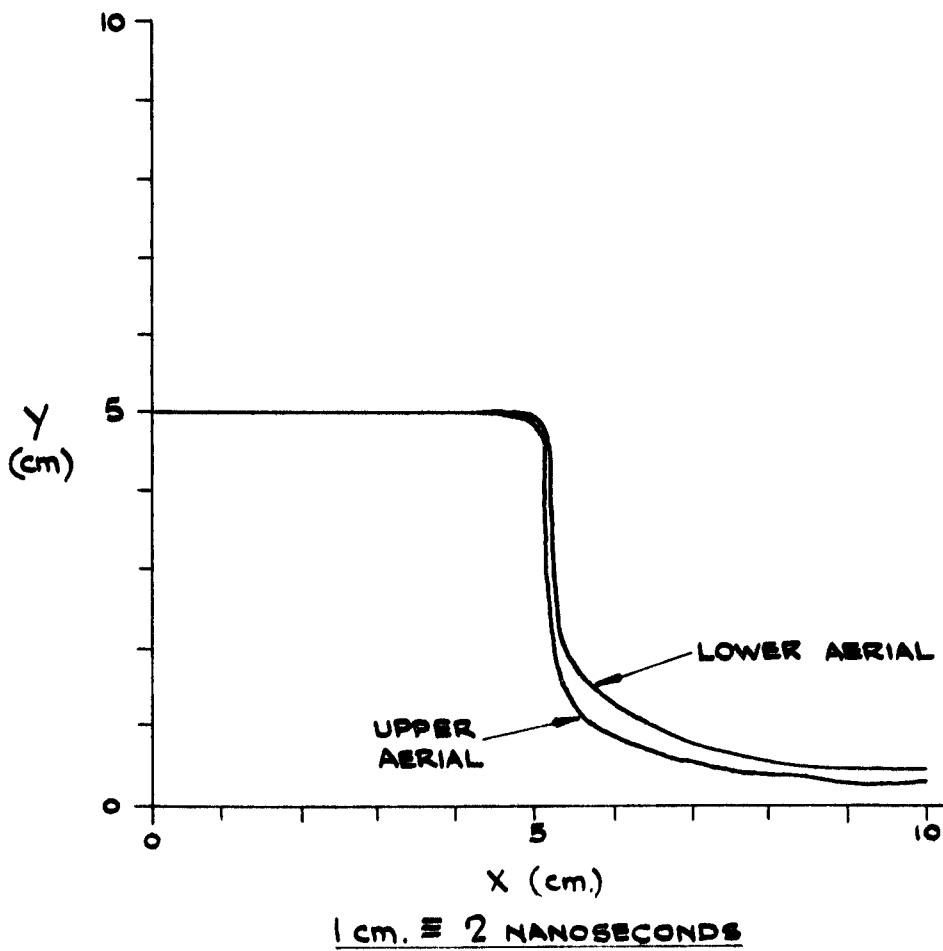
1 cm = 1 METRE OF CO-AXIAL CABLE.

SYSTEM 'S'

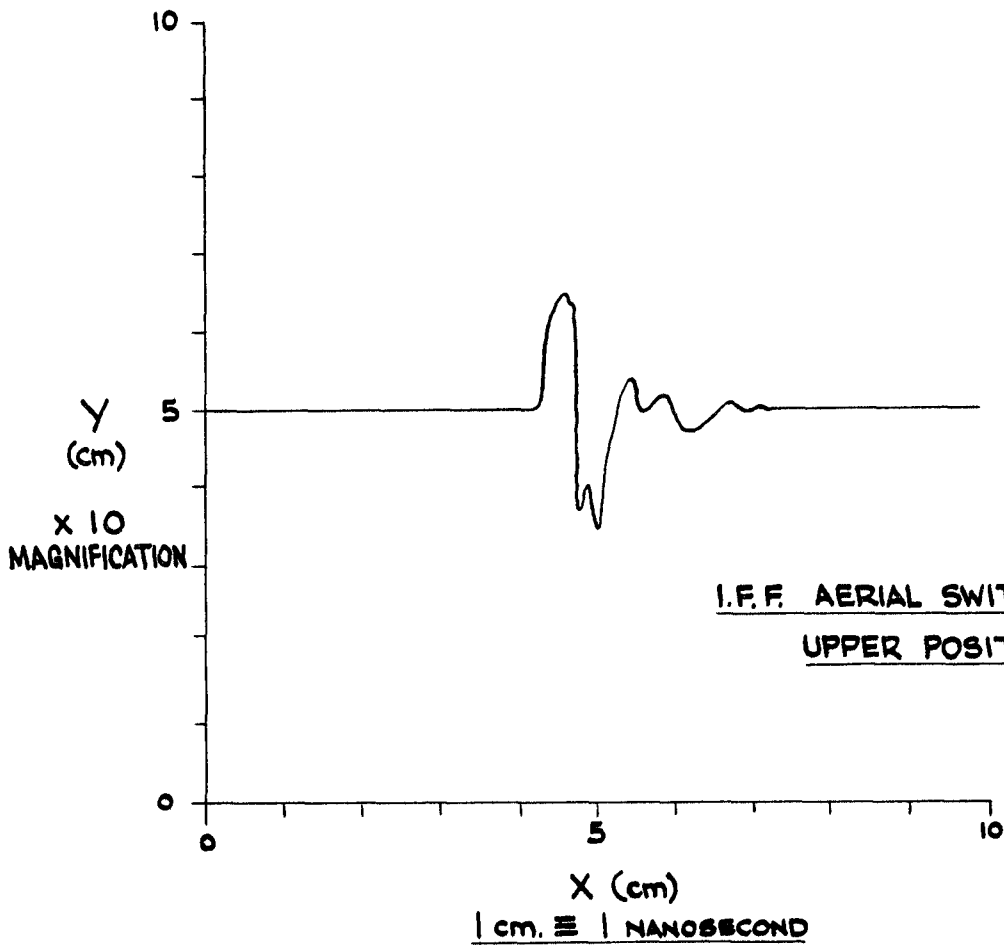
CANBERRA  
FIG. 2.



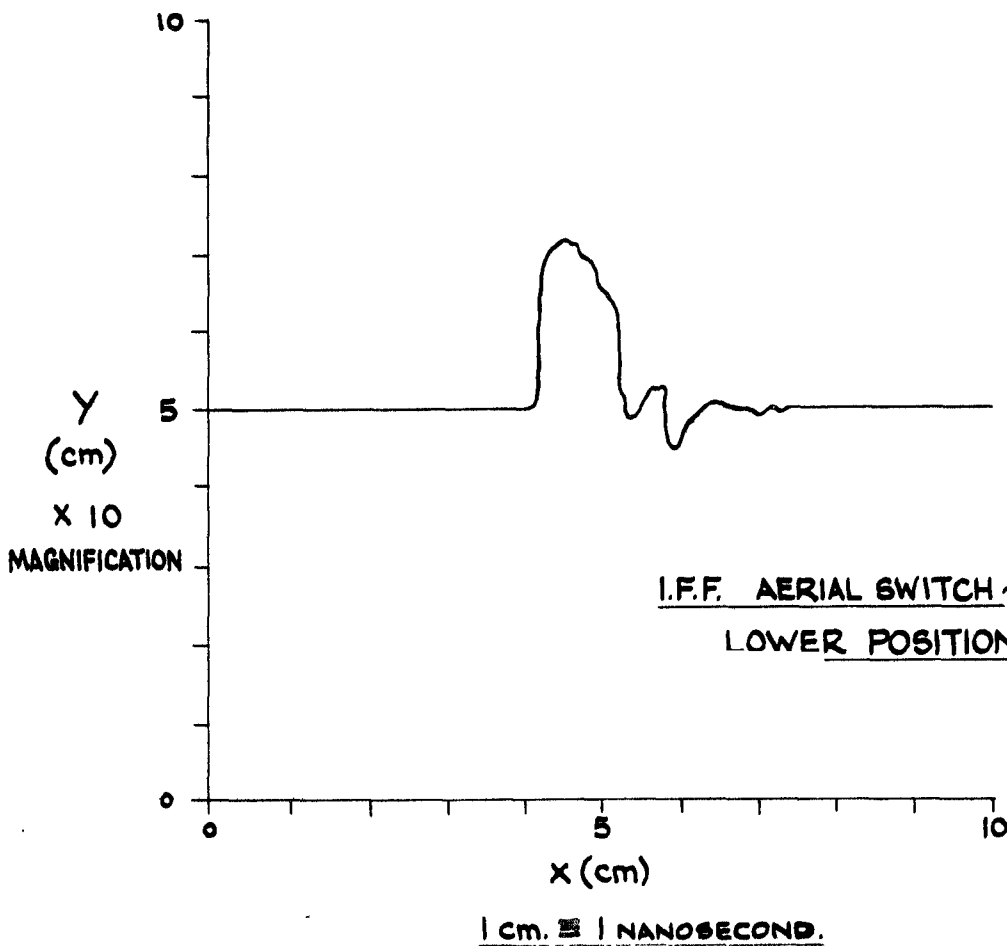
LIGHTNING F.M.K.3  
FIG. 3



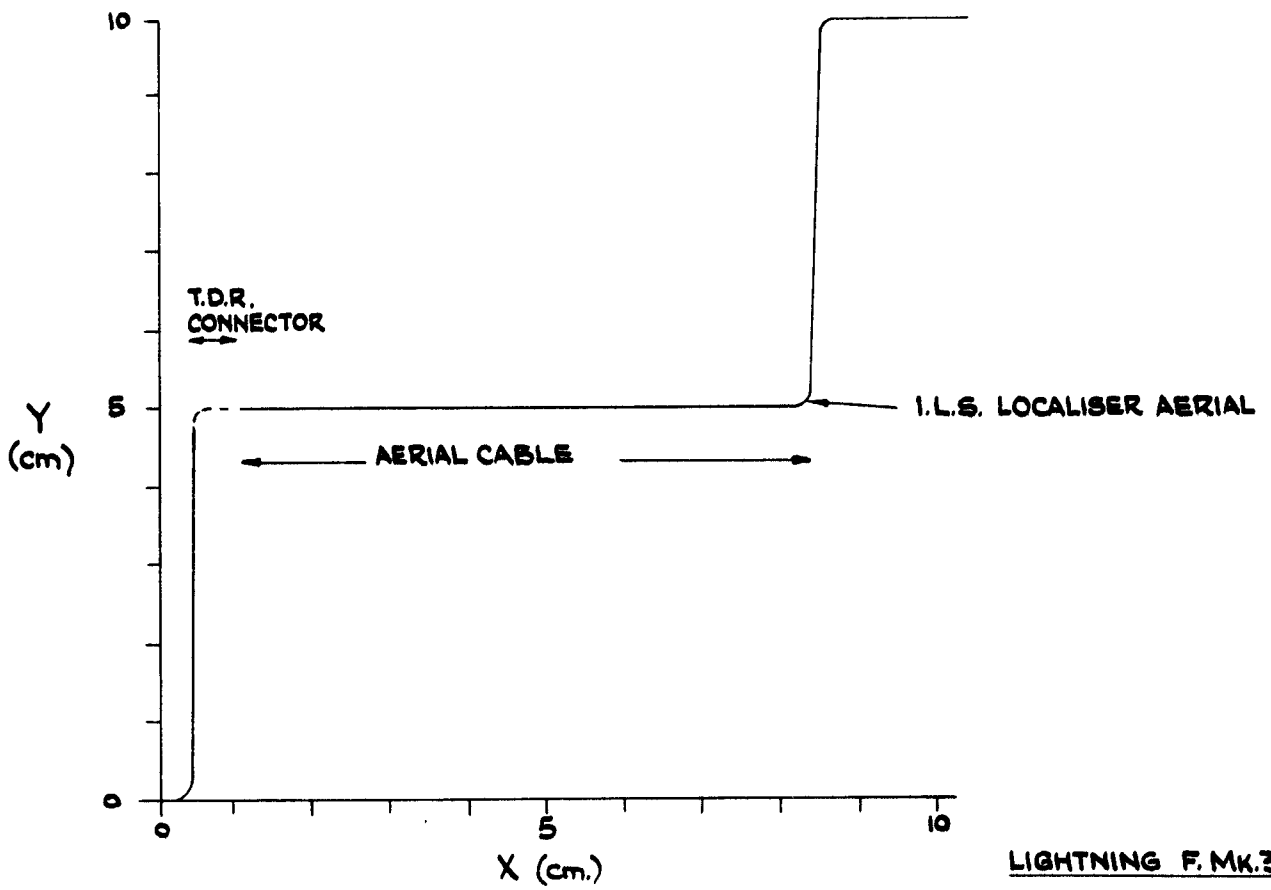
LIGHTNING F.M.K.3  
FIG. 4.



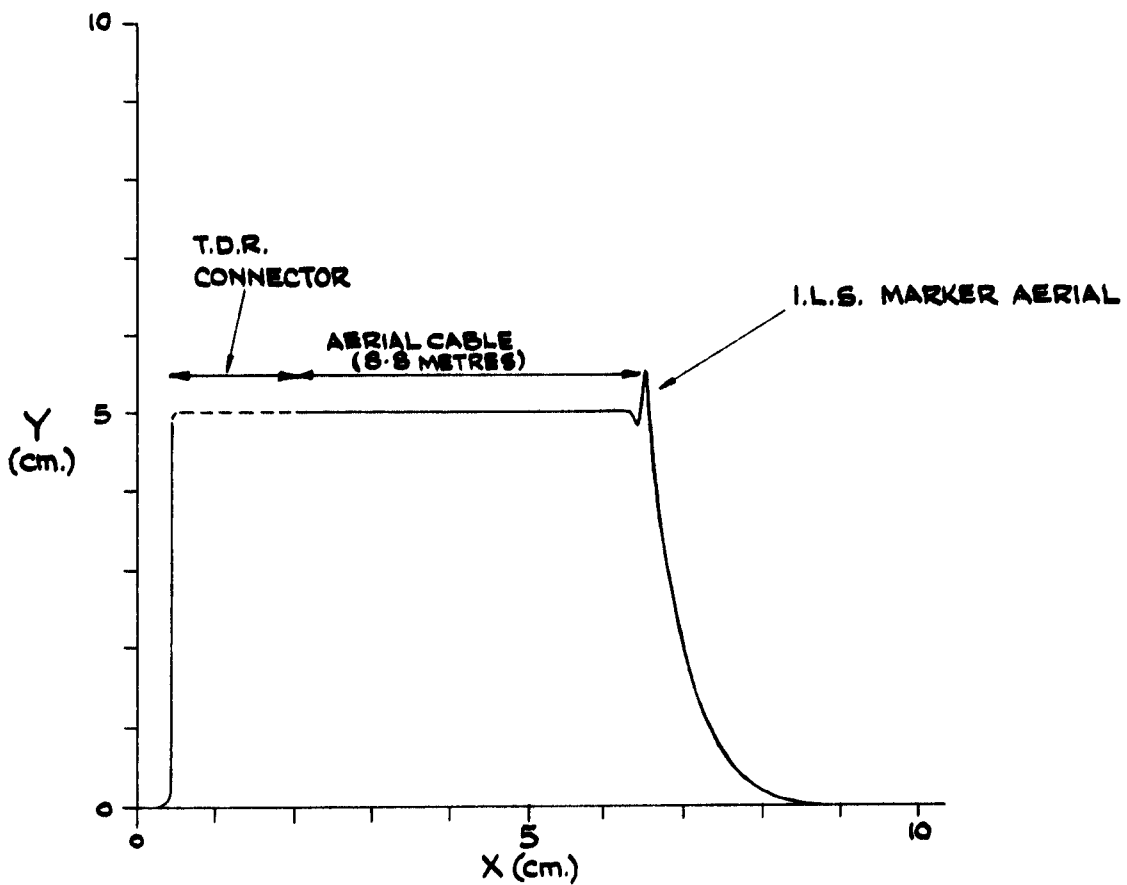
LIGHTNING F.MK.3  
FIG. 5A.



LIGHTNING F.MK.3.  
FIG. 5B.

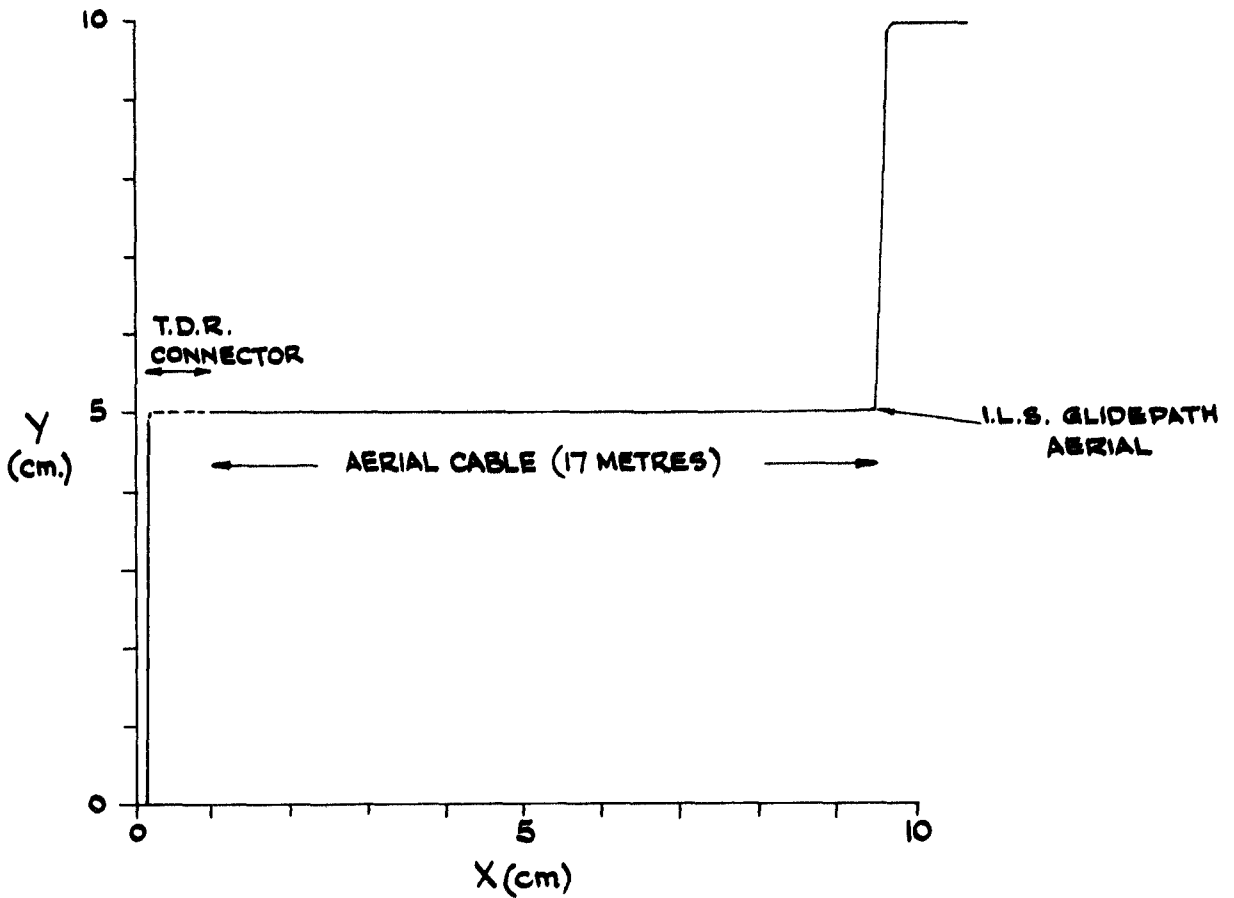


LIGHTNING F.MK.3  
FIG. 6



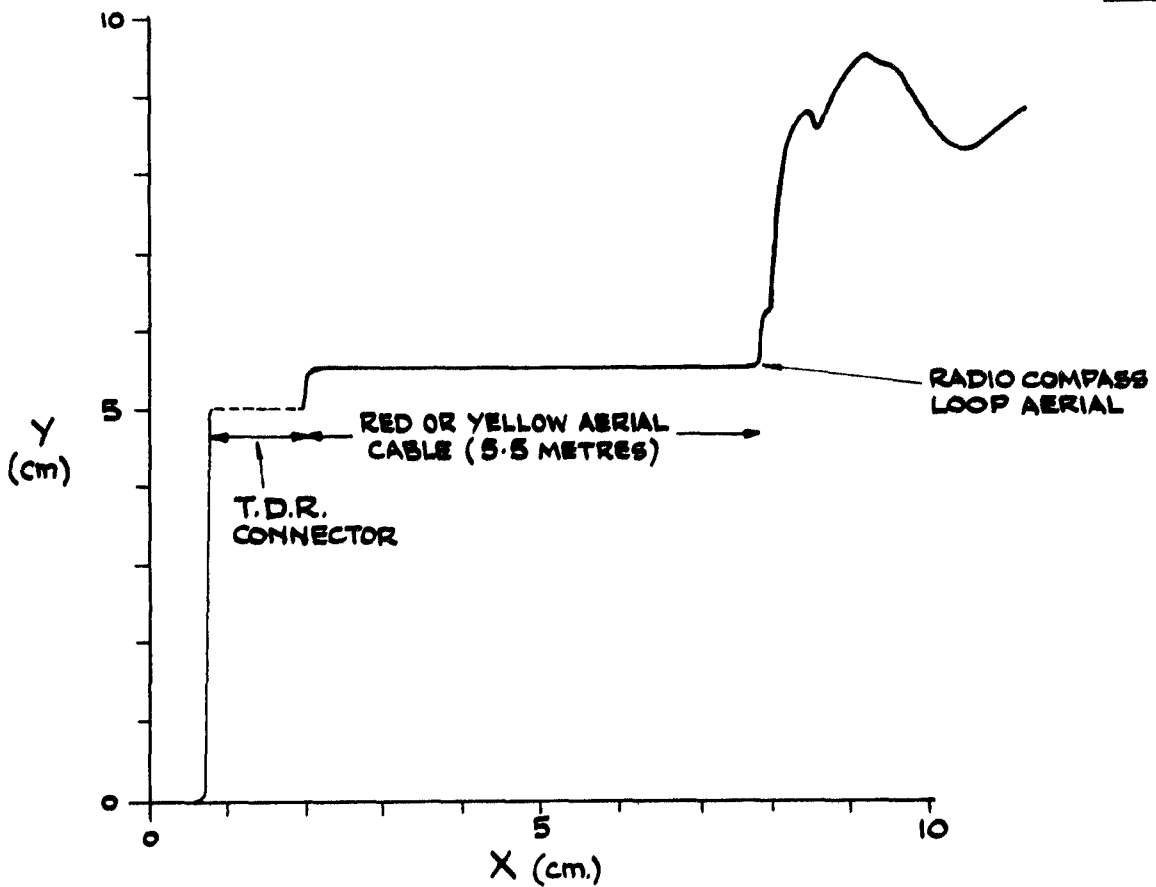
1 cm.  $\equiv$  2 METRES OF CO-AXIAL CABLE.

LIGHTNING F.MK.3  
FIG. 7.



1 cm.  $\equiv$  2 METRES OF CO-AXIAL CABLE.

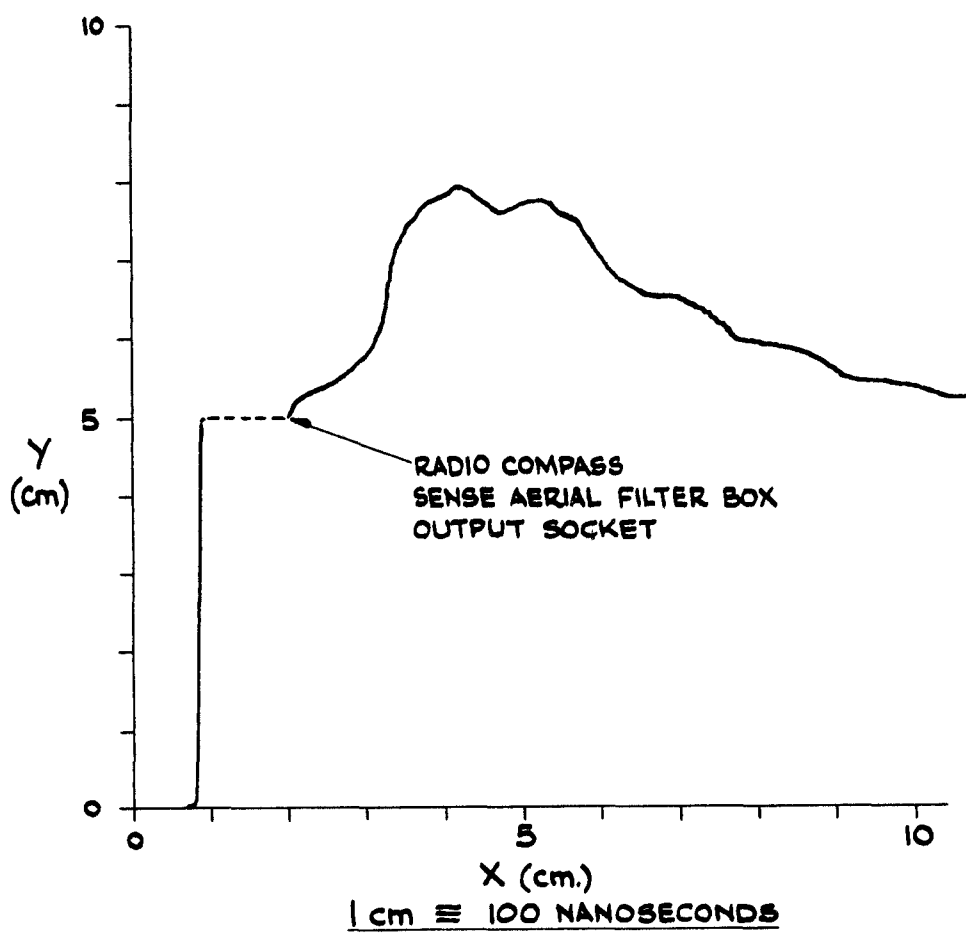
LIGHTNING F. Mk.3  
FIG. 8



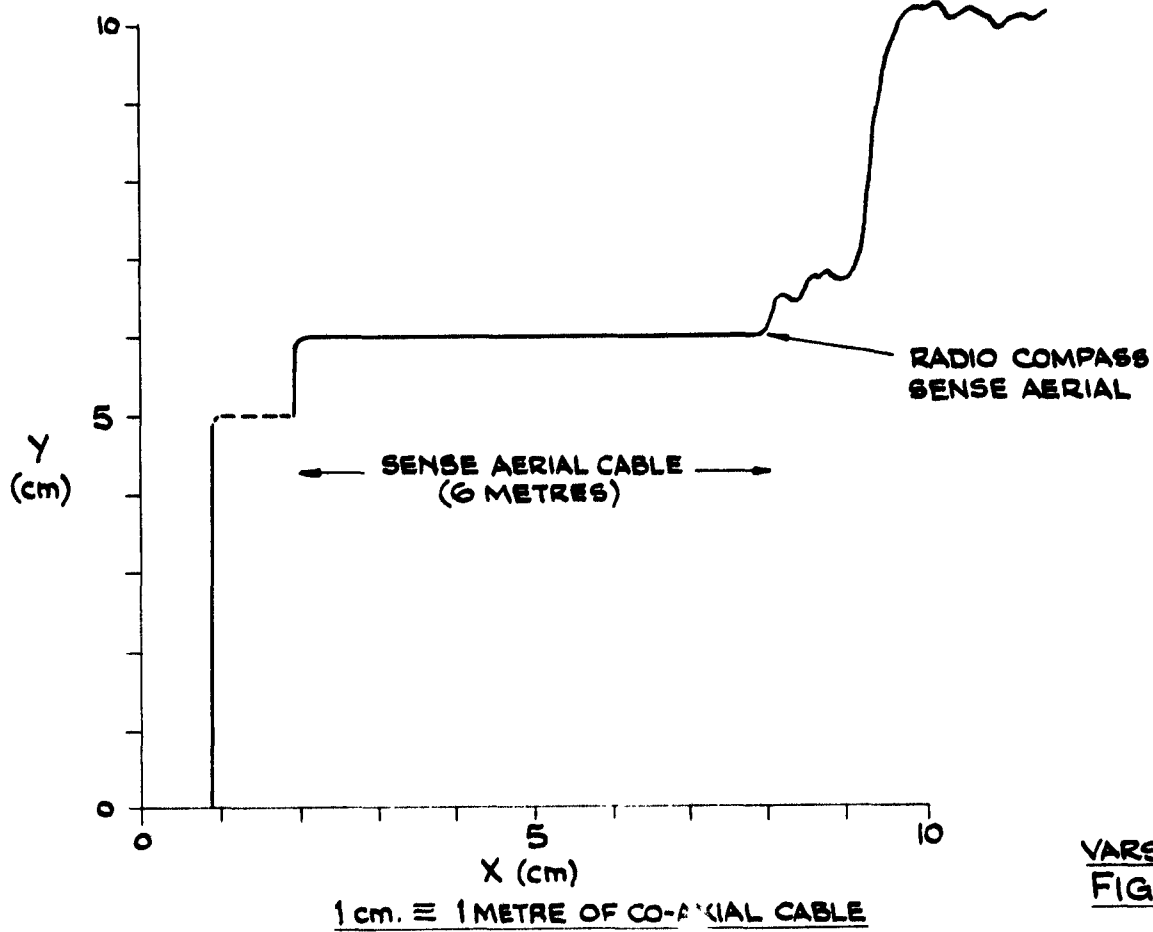
1 cm.  $\equiv$  1 METRE OF CO-AXIAL CABLE.

VARSITY  
FIG. 9



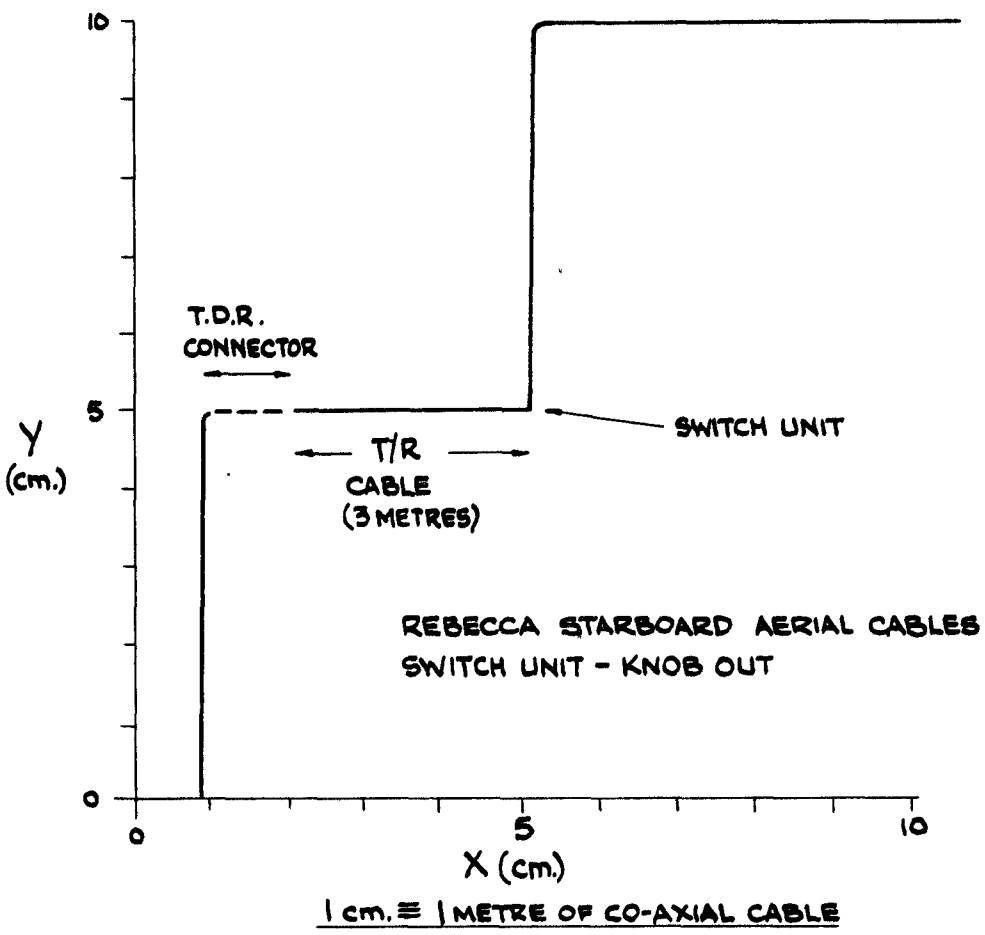


VARSIITY  
FIG. 10

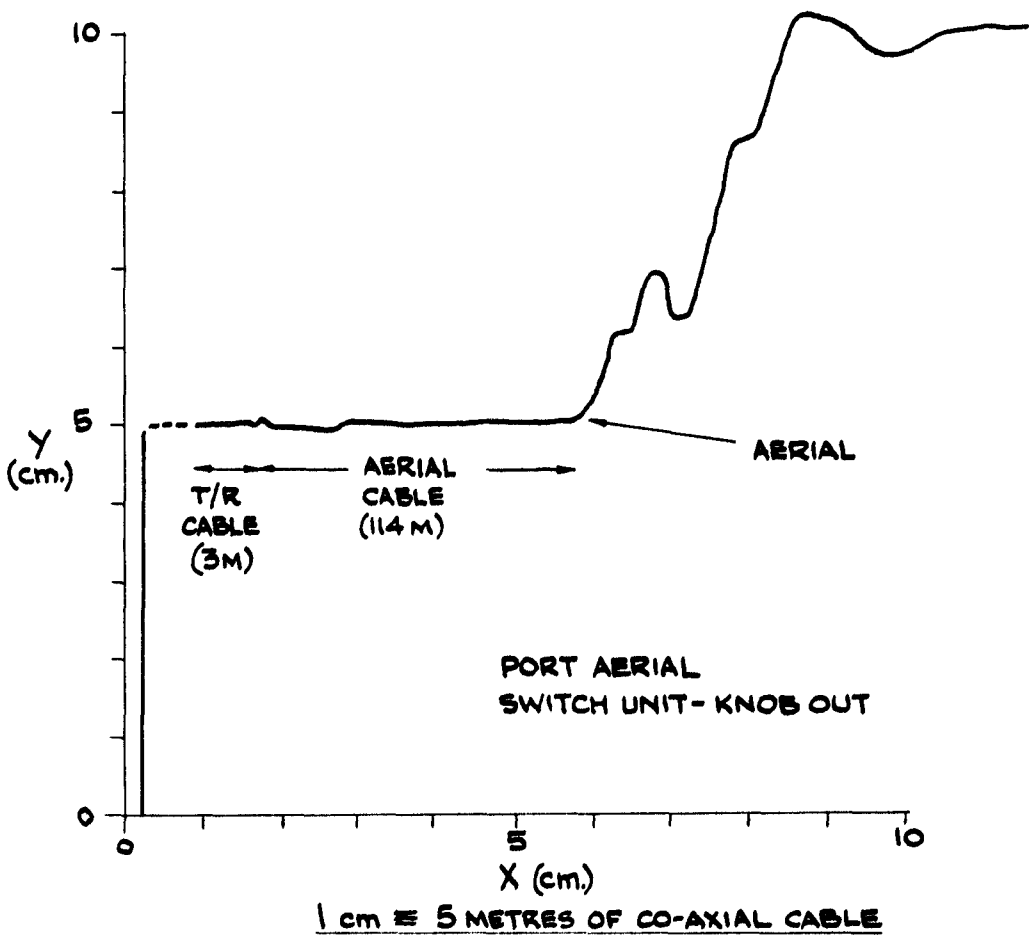


VARSIITY  
FIG. 11

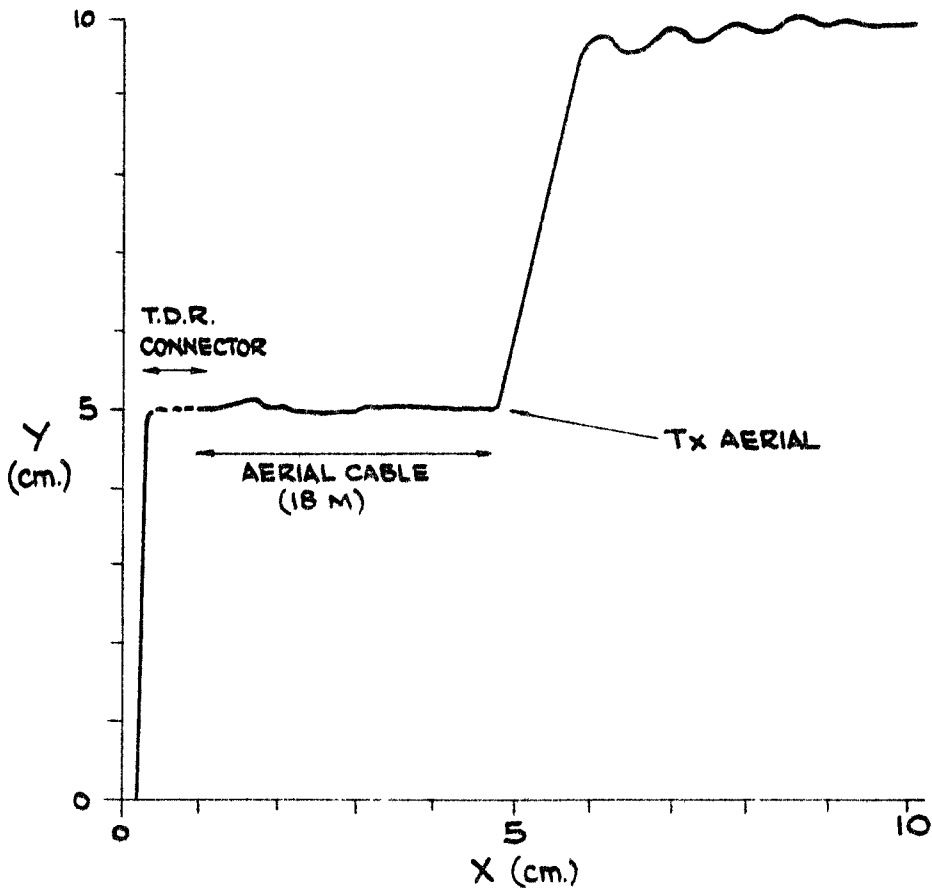




VARSBITY  
FIG. 14

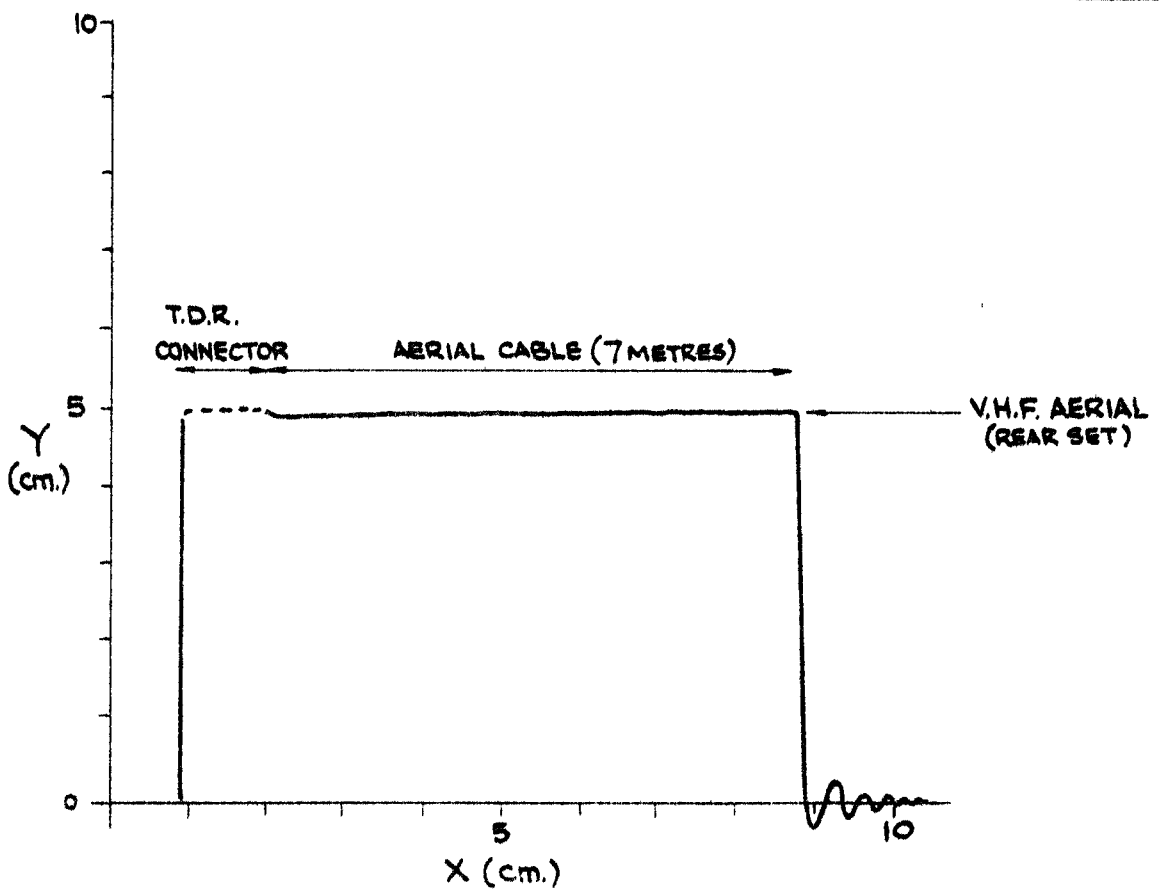


VARSBITY  
FIG. 15



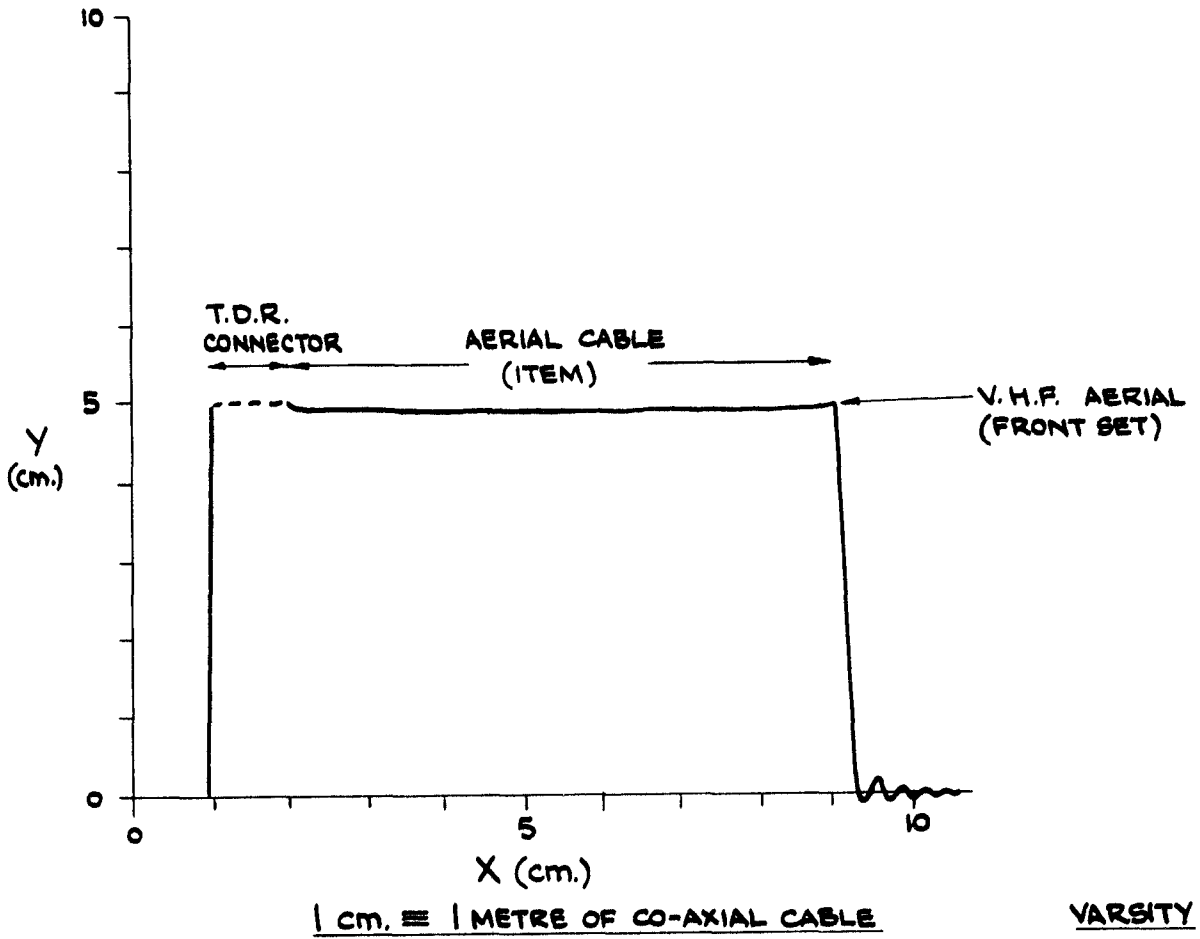
1 cm.  $\equiv$  5 METRES OF CO-AXIAL CABLE

VARSITY  
FIG. 16

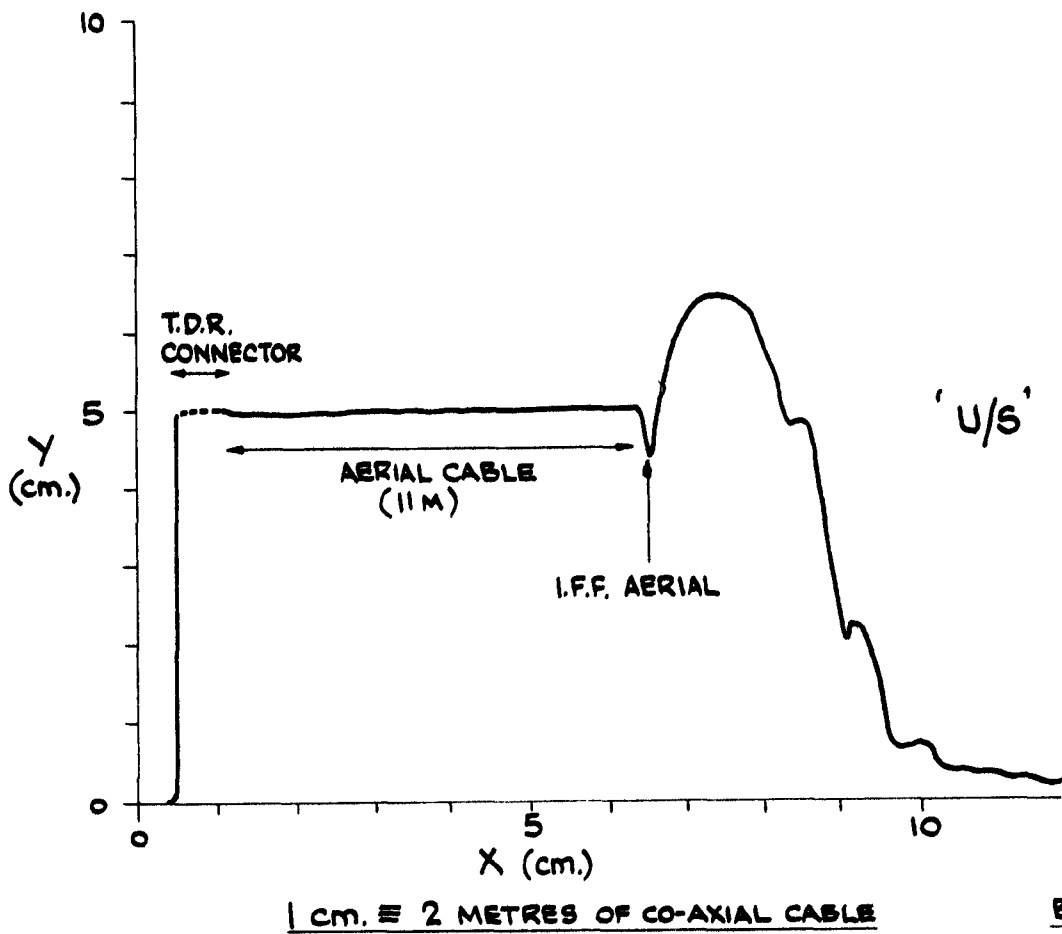


1 cm.  $\equiv$  1 METRE OF CO-AXIAL CABLE

VARSITY  
FIG. 17

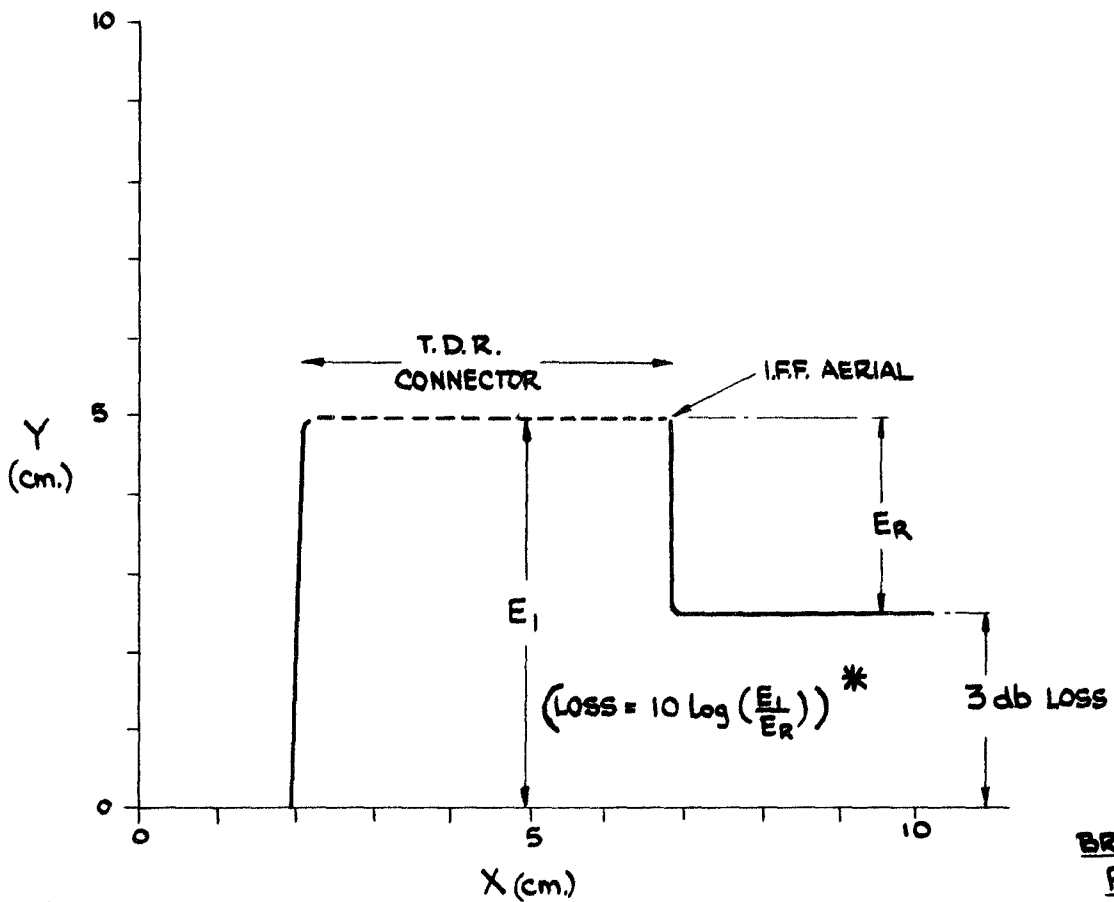


VARSIITY  
FIG. 18



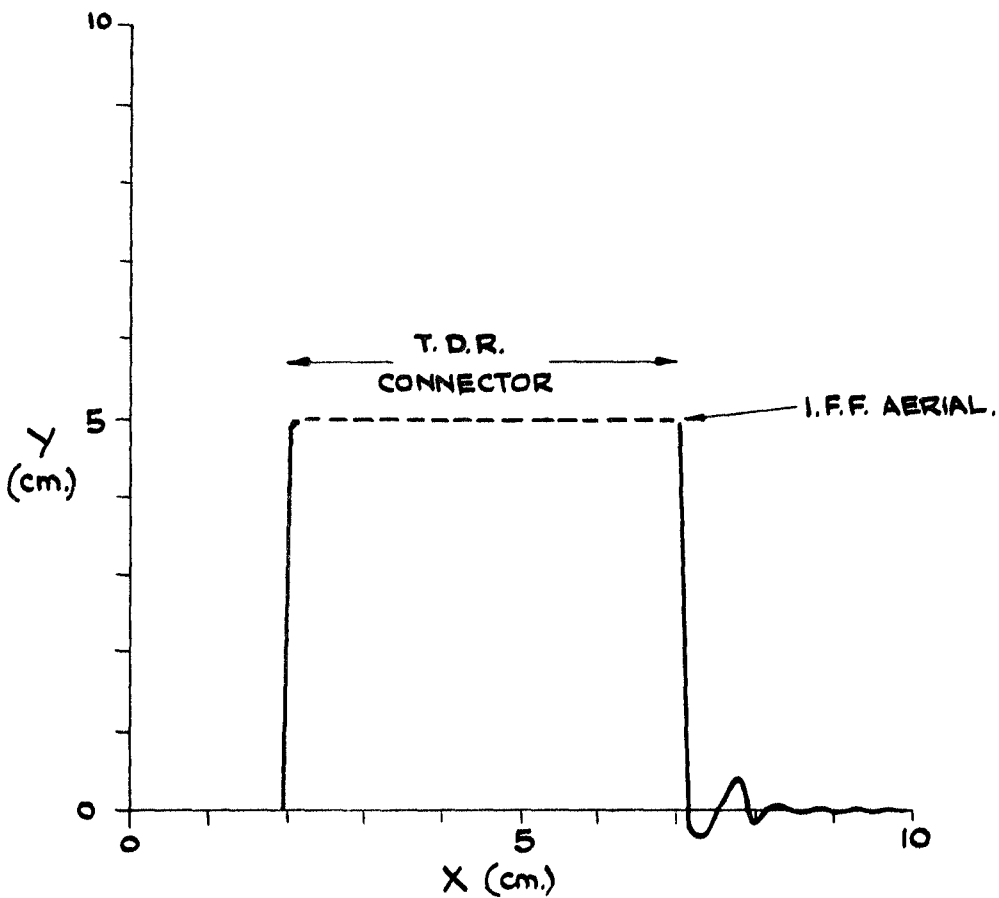
BRITANNIA  
FIG. 19





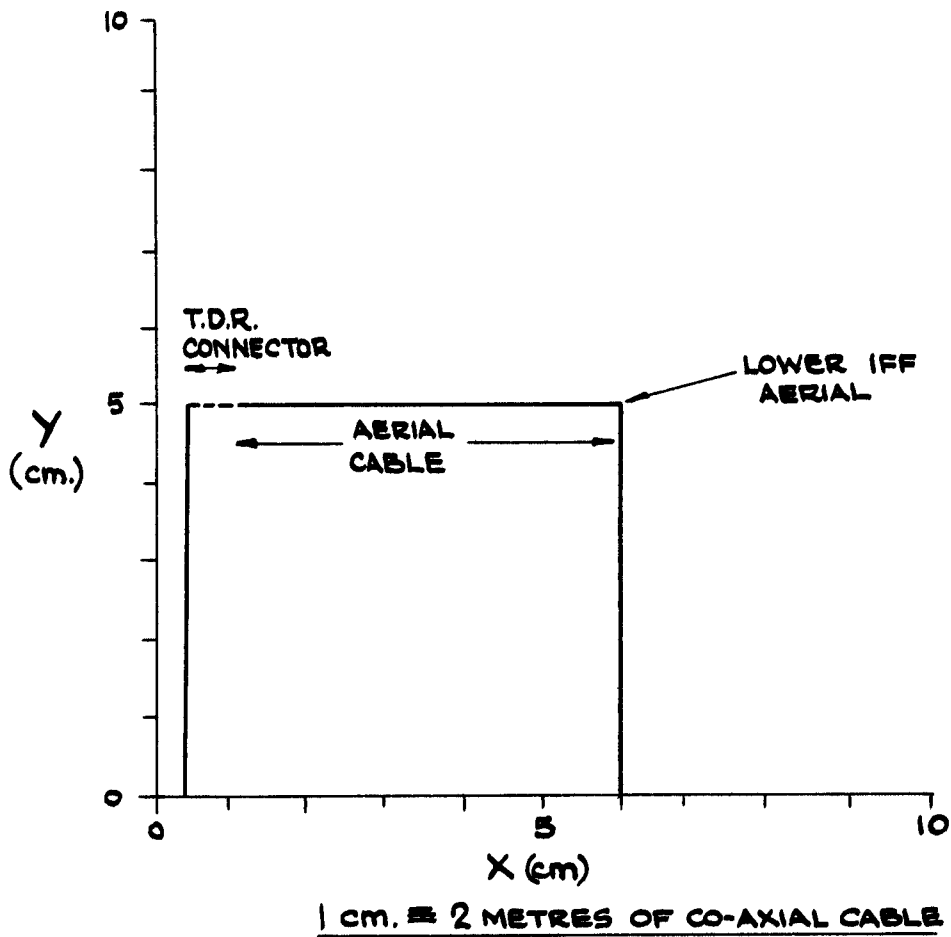
BRITANNIA  
FIG. 20

\* BROAD-BAND ATTENUATION LOSSES CAN ONLY BE MEASURED DIRECTLY WITH OPEN OR SHORT CIRCUIT TERMINATIONS. (FIG. 20 - SHORT CCT.)

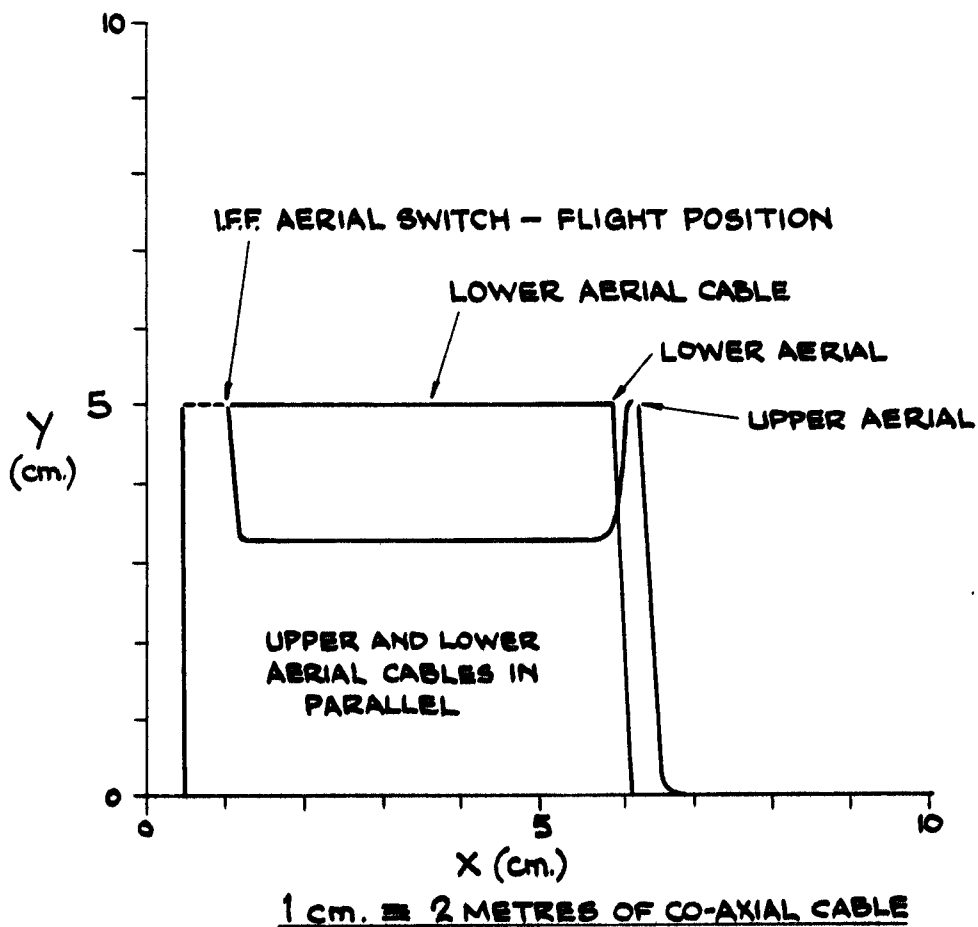


1 cm. = 2 METRES OF CO-AXIAL CABLE.

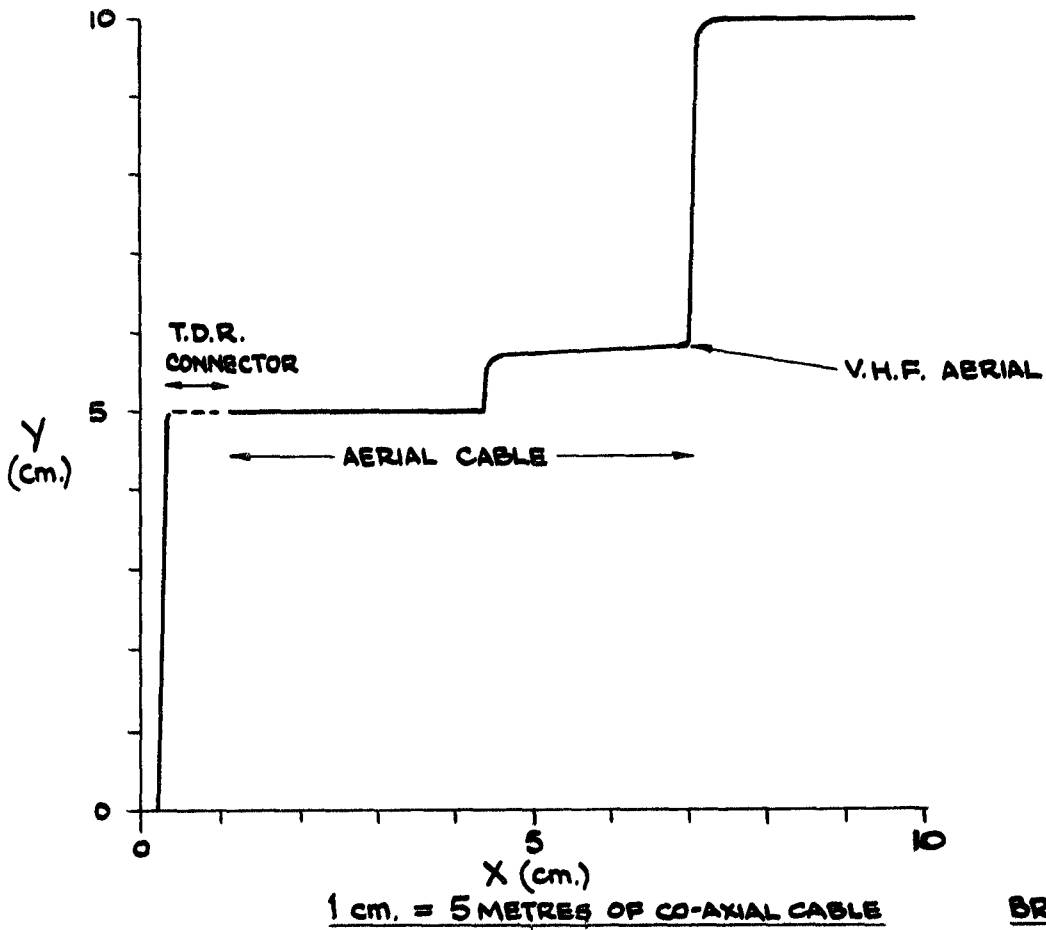
BRITANNIA  
FIG. 21



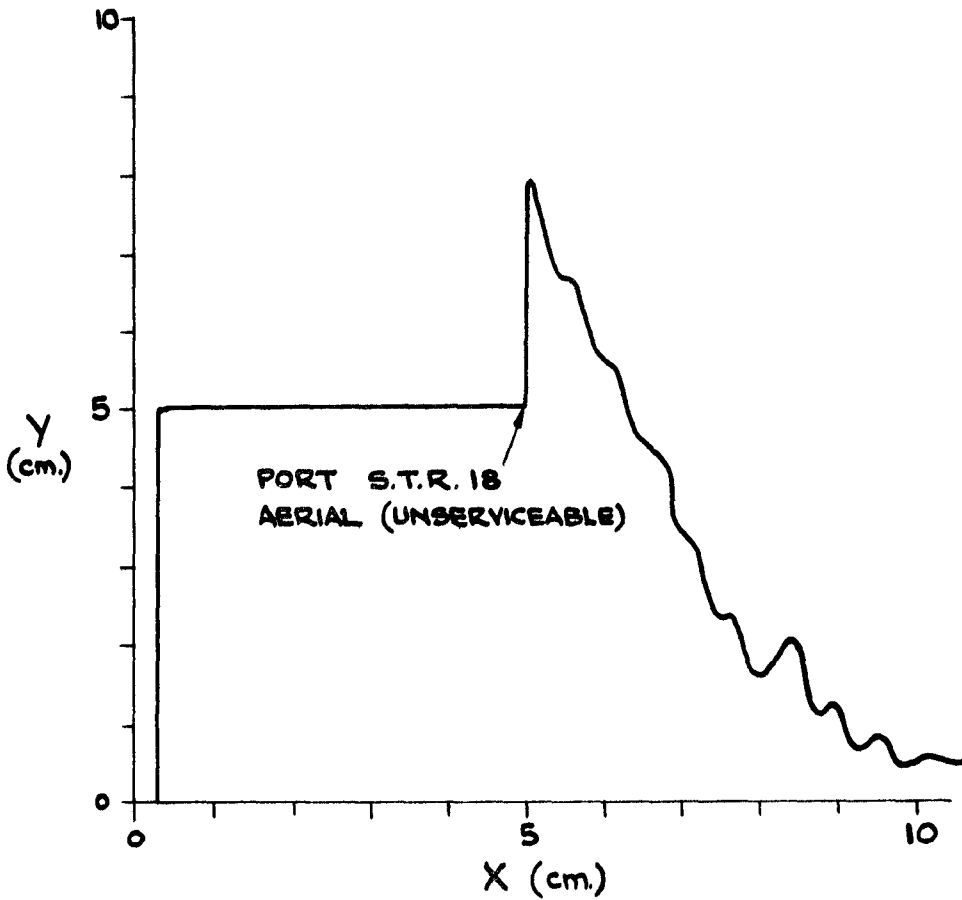
BRITANNIA  
FIG. 22



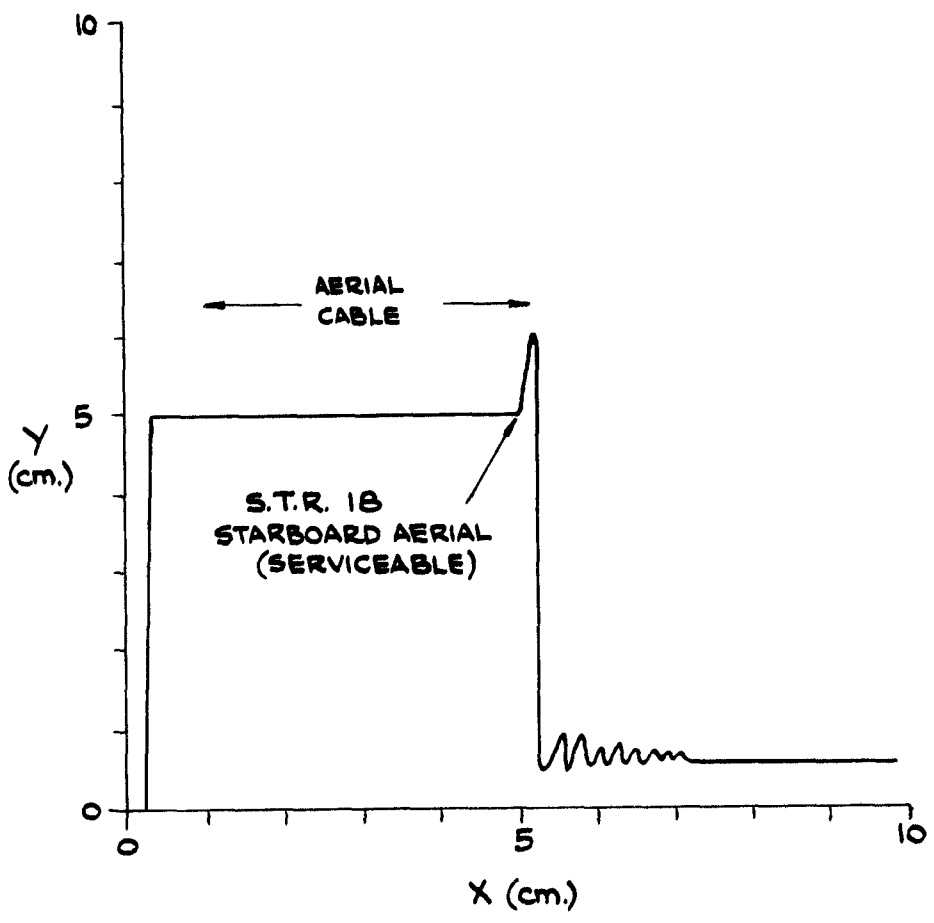
BRITANNIA  
FIG. 23



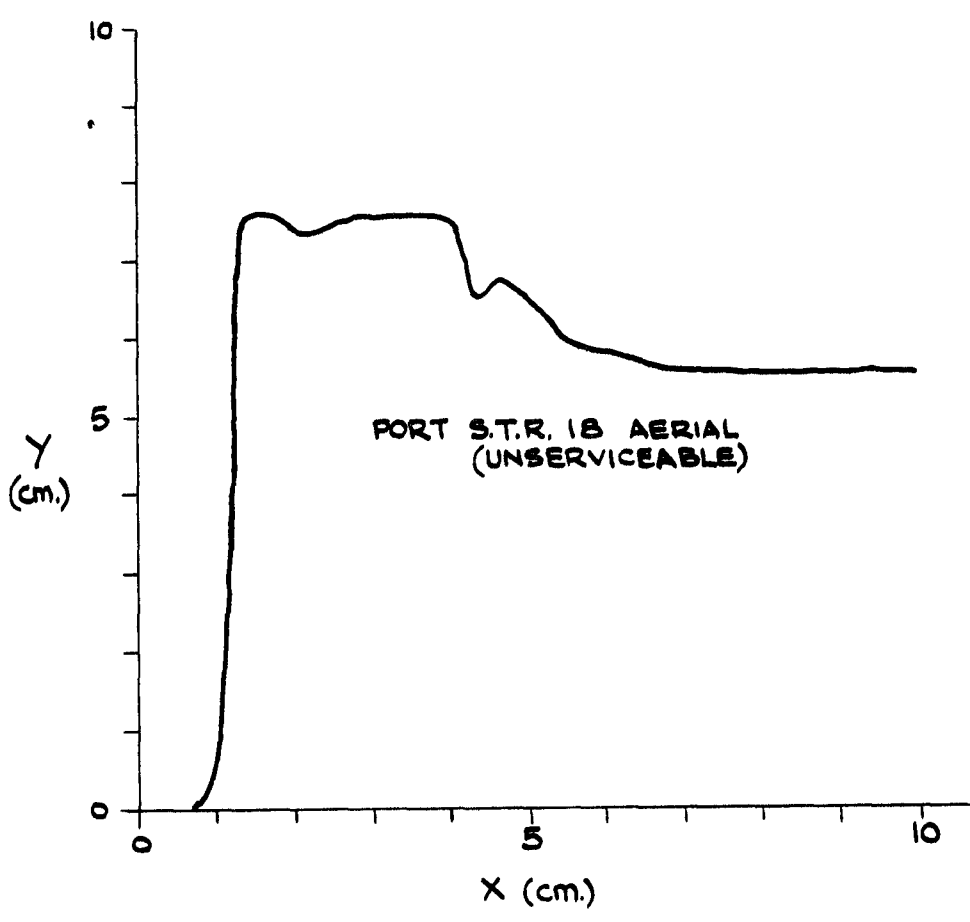
BRITANNIA  
FIG. 24



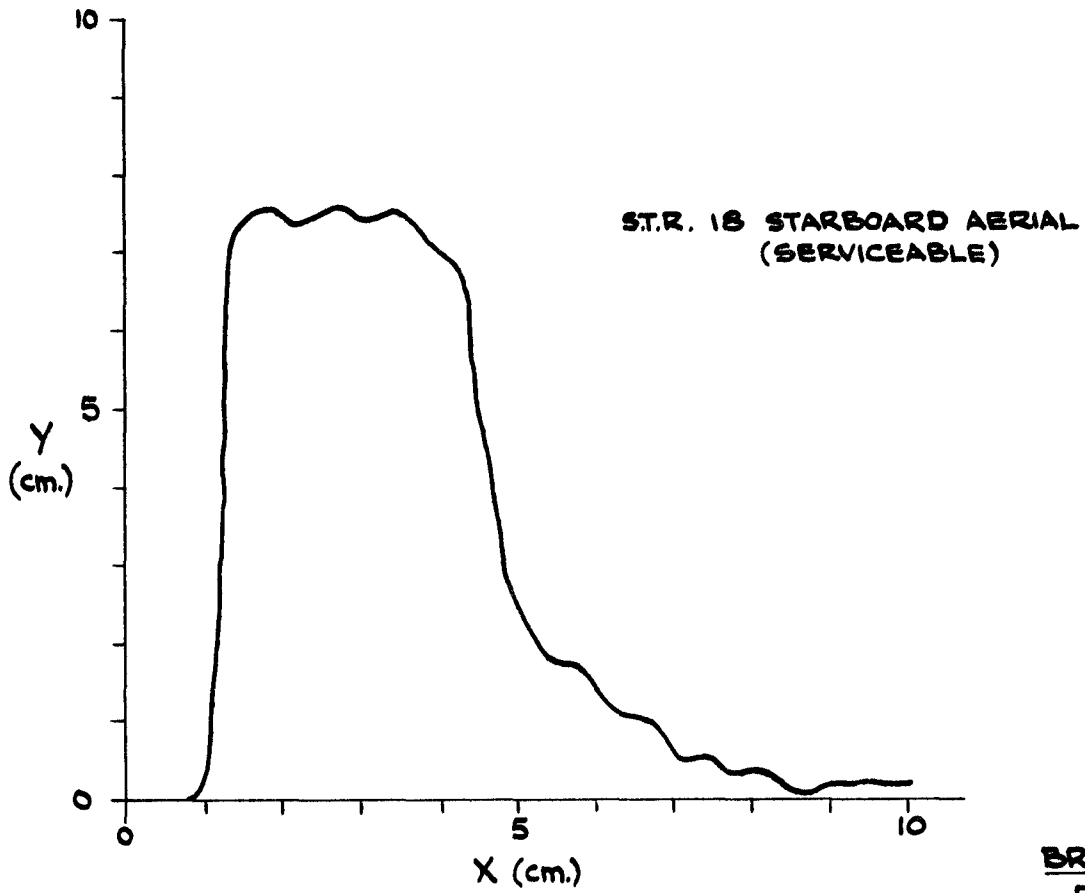
BRITANNIA  
FIG. 25



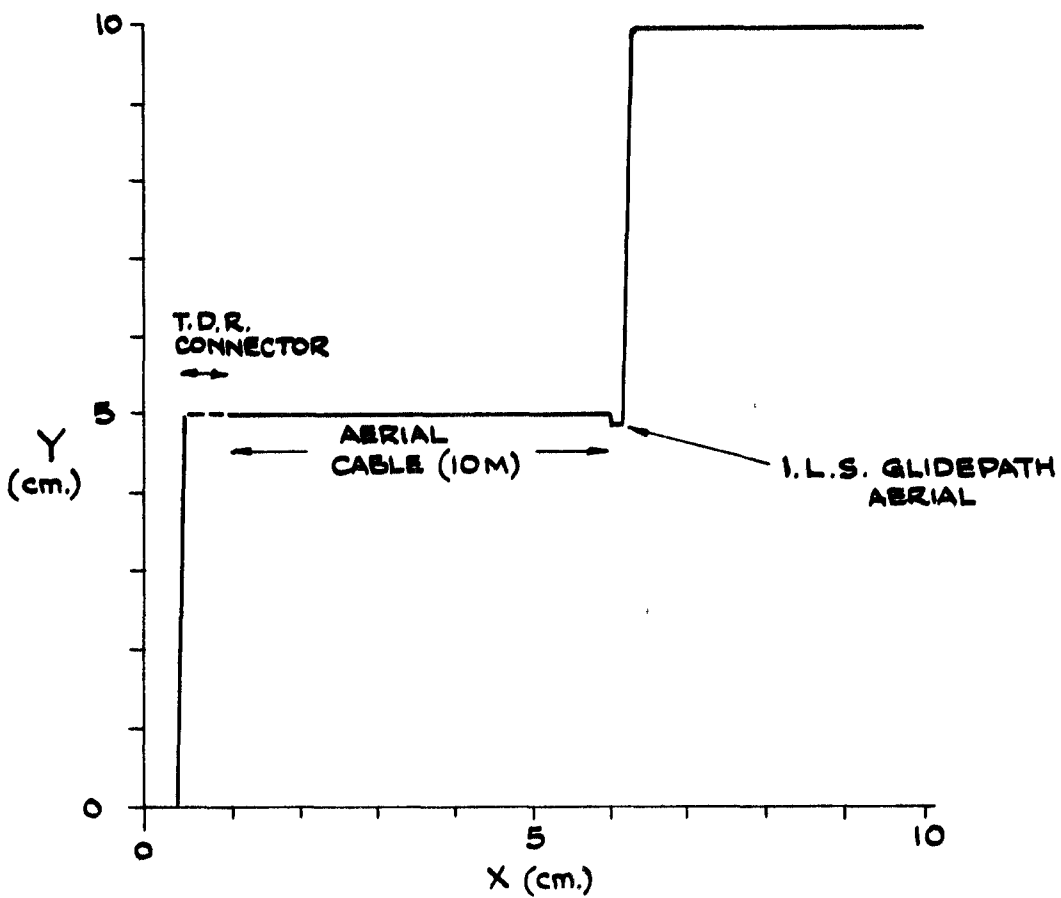
BRITANNIA  
FIG. 26



BRITANNIA  
FIG. 27.

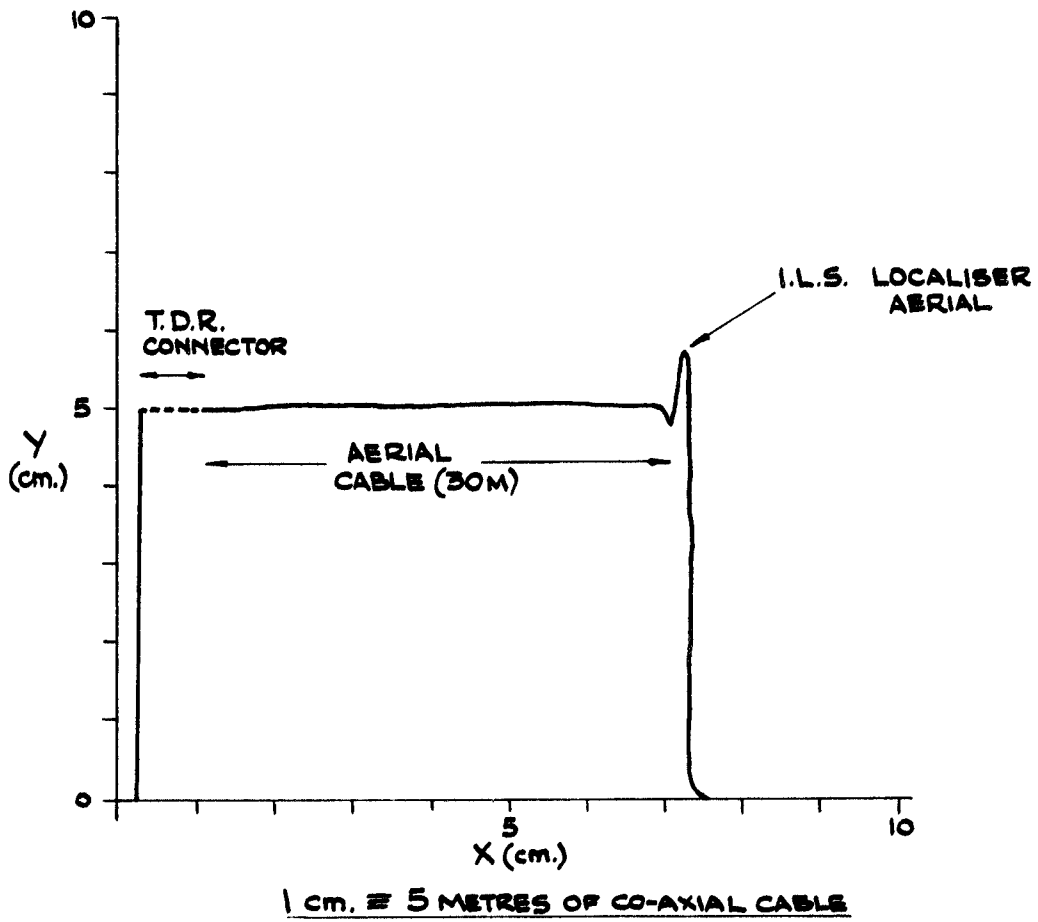


BRITANNIA  
FIG. 28

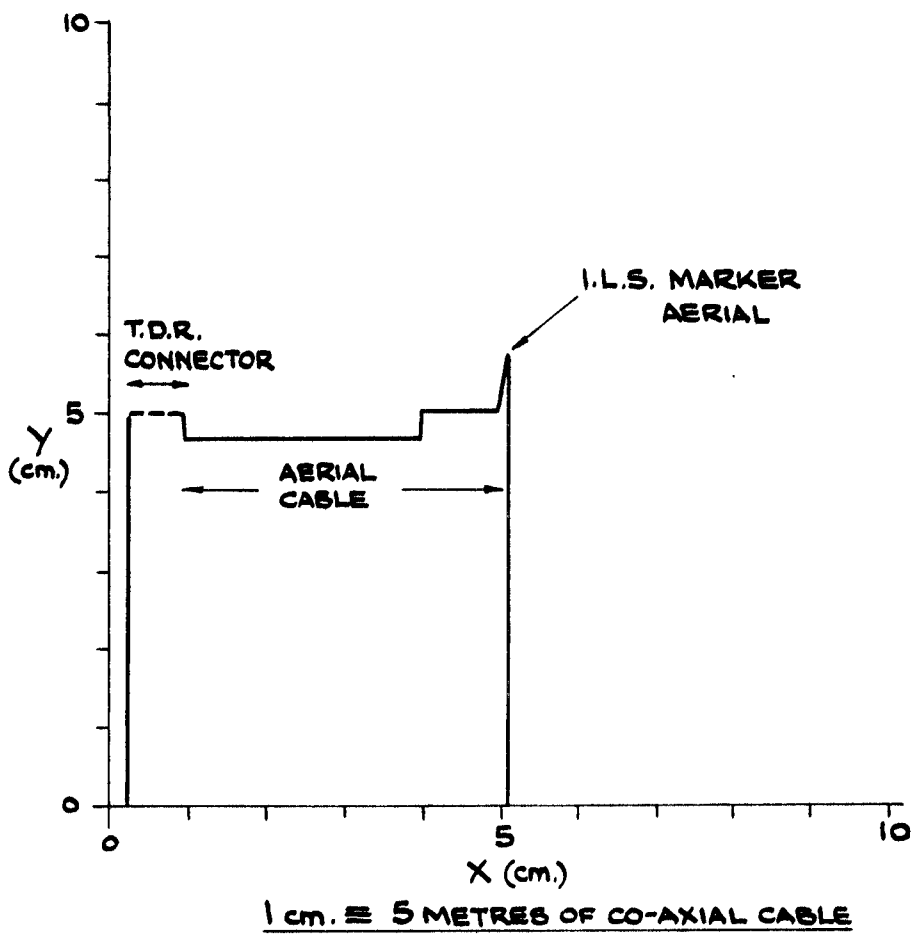


1 cm. = 2 METRES OF CO-AXIAL CABLE

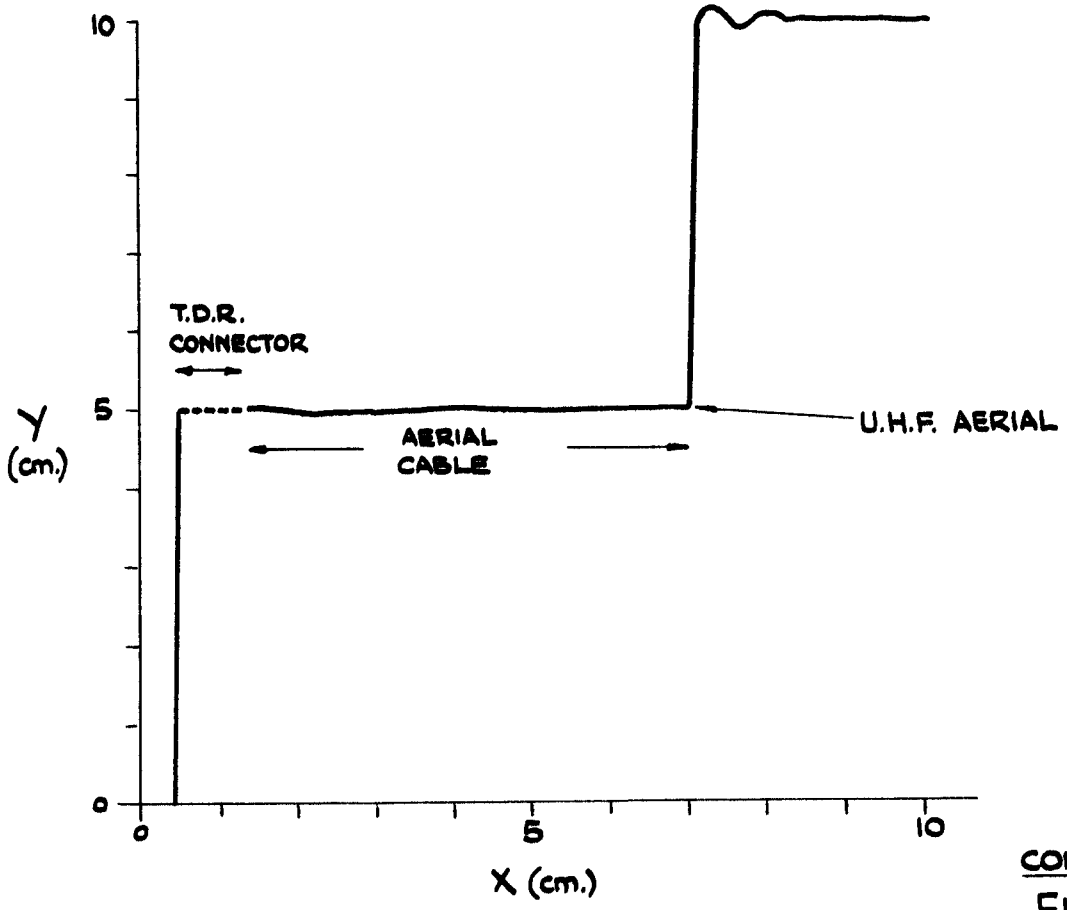
COMET  
FIG. 29



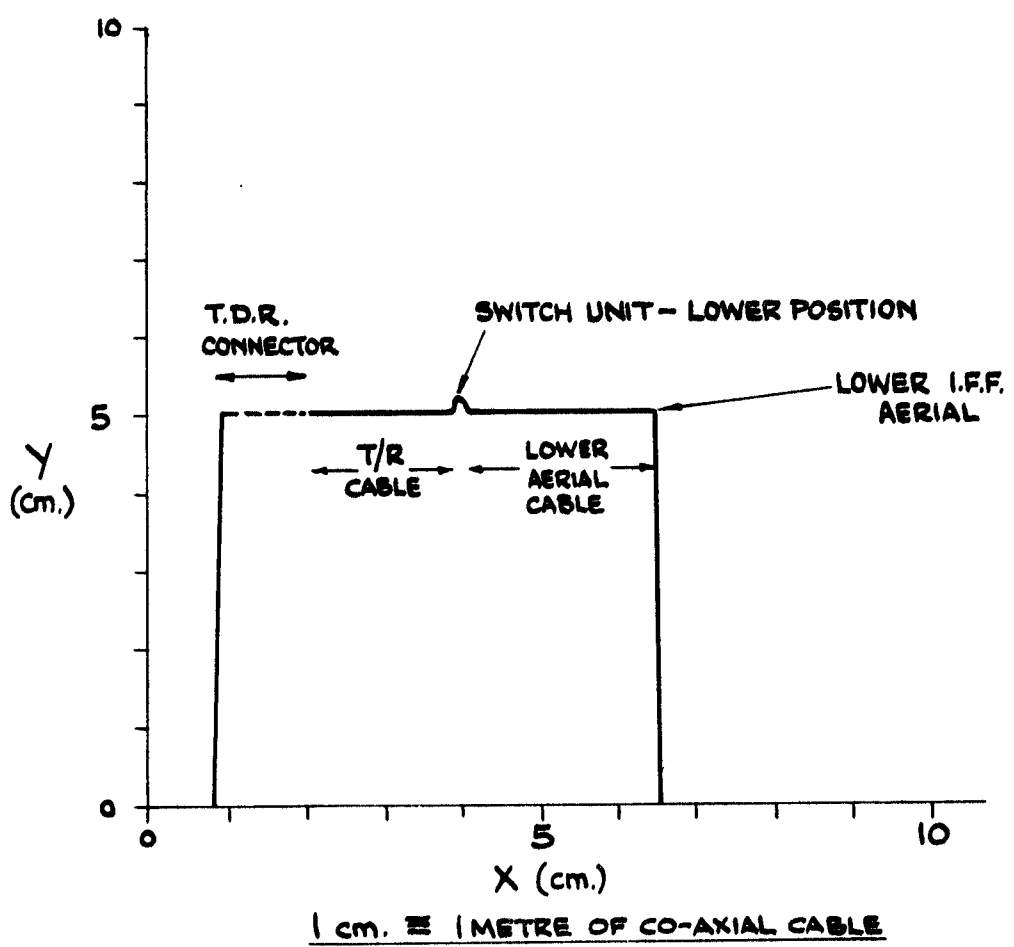
COMET  
FIG.30



COMET  
FIG.31.

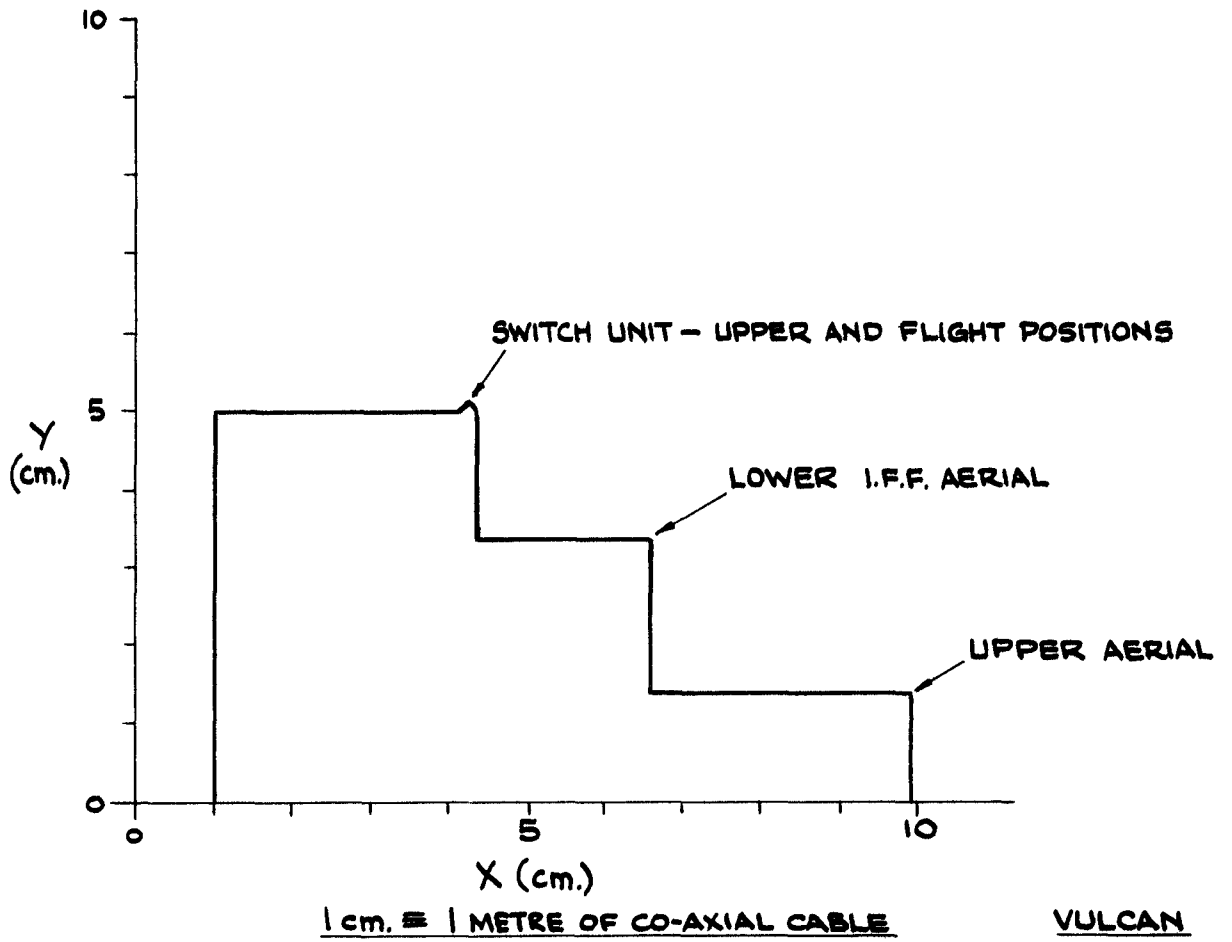


**COMET**  
**FIG.32**

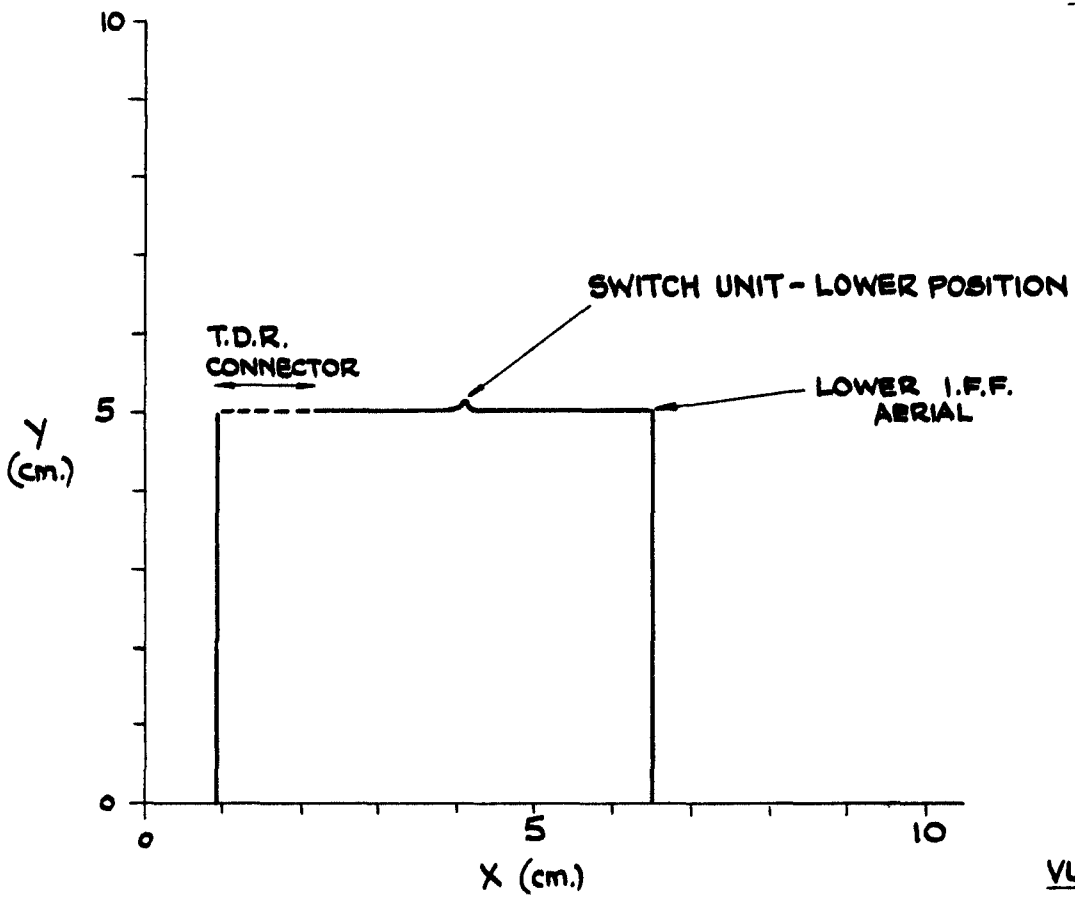


1 cm. ≡ 1 METRE OF CO-AXIAL CABLE

**VULCAN**  
**FIG.33**

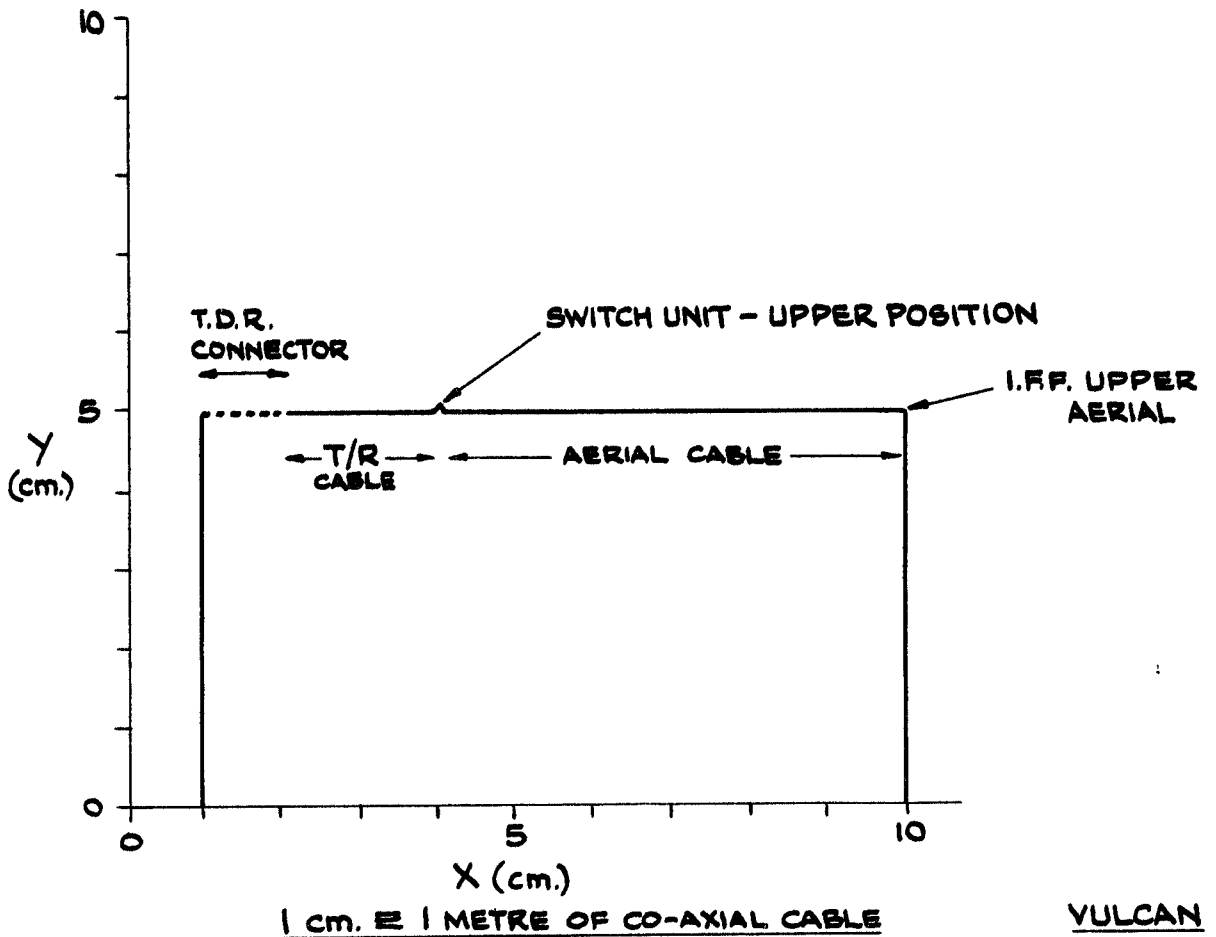


VULCAN  
FIG. 34

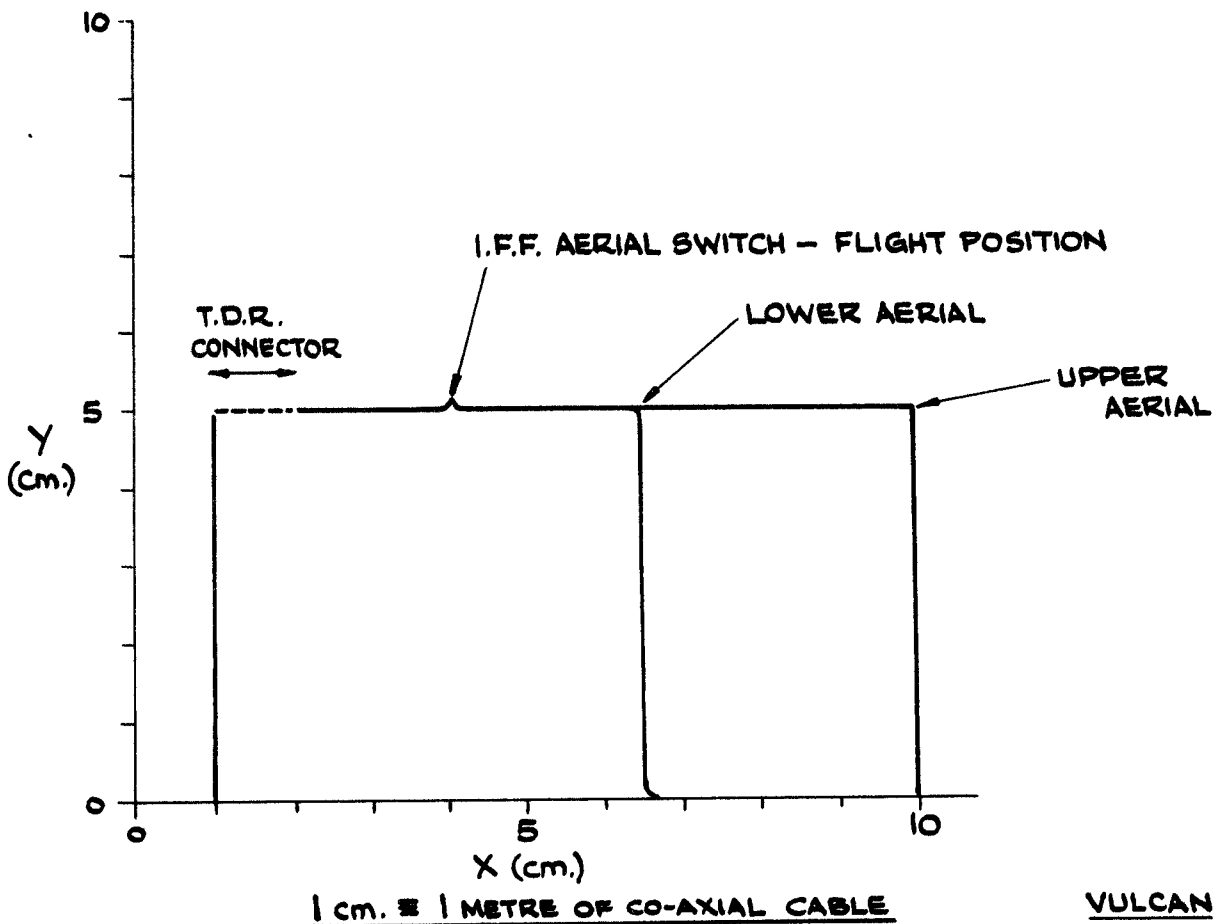


VULCAN  
FIG. 35





VULCAN  
FIG.36



VULCAN  
FIG.37